

# TEC Evaluation from GNSS Measurements

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Inspired by Gigi (Gg) Ciruolo



**ISTITUTO NAZIONALE DI GEOFISICA E VULCANOLOGIA**

**African Capacity Building Workshop on Space Weather Effects on GNSS**

**03 - 14 October 2022**

**Trieste, Italy**

# What did we learn about GNSS and TEC so far?

A lot of cool stuff but mainly...

- There are a lot of GNSS satellites and receivers available nowadays
- GNSS transmit signals on (at least) 2 different frequencies
- We can get code and phase observables from GNSS signals
- Ionosphere impacts on the propagation of GNSS signals from the satellite to the receiver
- We can use GNSS to derive TEC

**100% GOOD STUFF**

But...

- It is not straightforward to measure TEC by means of GNSS signals
- There are “biases” affecting TEC measurements from GNSS



# Goals to be reached in the next hour (or less if get bored)

- Understand how to use GNSS observables to retrieve TEC
- Understand what problems we have to face to do that
- Have a rough idea on how the problems can be solved (one of the possible solutions)
- Have the possibility to run a software capable of estimating TEC from GNSS measurements



# Outline

Ionosphere refractive index

Total Electron Content

Geometry-free linear combination of GNSS observables

TEC calibration and mapping

Ciraolo (Gigi) calibration technique

Gigi's software

Live session (if we are on time)

# The Appleton equation

To describe the refractive index of the ionosphere,  $n$ , we can refer to the Appleton equation:

$$n^2 = 1 - \frac{X}{1 - iZ - \frac{Y_T^2}{2(1 - X - iZ)} \pm \sqrt{\frac{Y_T^2}{4(1 - X - iZ)^2 + Y_L^2}}}$$

$$X = \frac{\omega_0^2}{\omega^2}$$

$$Y = \frac{\omega_H}{\omega}$$

$$Z = \frac{\nu}{\omega}$$

$$\omega = 2\pi f$$

Electron plasma frequency

Electron gyro frequency

Collision frequency

$$\omega_0 = \sqrt{\frac{Ne^2}{\epsilon_0 m}}$$

$$\omega_H = \frac{Be}{m}$$

$\nu$  Can be neglected in most of the case



No Absorption

$$B = 0 \quad \rightarrow \quad n = \sqrt{1 - X}$$

# The Appleton equation (GNSS frequencies)

The frequency of signals used for positioning has to be selected in order to make the refractive index as close as possible to unity (compatibly with international rules and status of art of technology).

Consider  $N = 10^{12} \text{ e/m}^3$  (a rather strong value) and  $f = 1.5 \text{ GHz}$  (representative of GNSS frequencies)

$$n \approx 1 - (2 * 10^{-5})$$

For frequencies used in positioning, it can be used a first order approximation of the Appleton-Hartree formula

$$X = \frac{\omega_0^2}{\omega^2} \ll 1 \quad Y = \frac{\omega_H}{\omega} \approx 0 \quad Z = \frac{v}{\omega} \approx 0$$

$$n^2 = 1 - \frac{X}{1 - iZ - \frac{Y^2}{2(1 - X - iZ)} \pm \sqrt{\frac{Y^2}{4(1 - X - iZ)^2 + Y^2}}} \quad \rightarrow \quad n = \sqrt{1 - X} \approx 1 - \frac{X}{2} = 1 - \frac{N_e e^2}{\epsilon_0 m} = 1 - \frac{40.3 N_e}{f^2}$$

Electron density

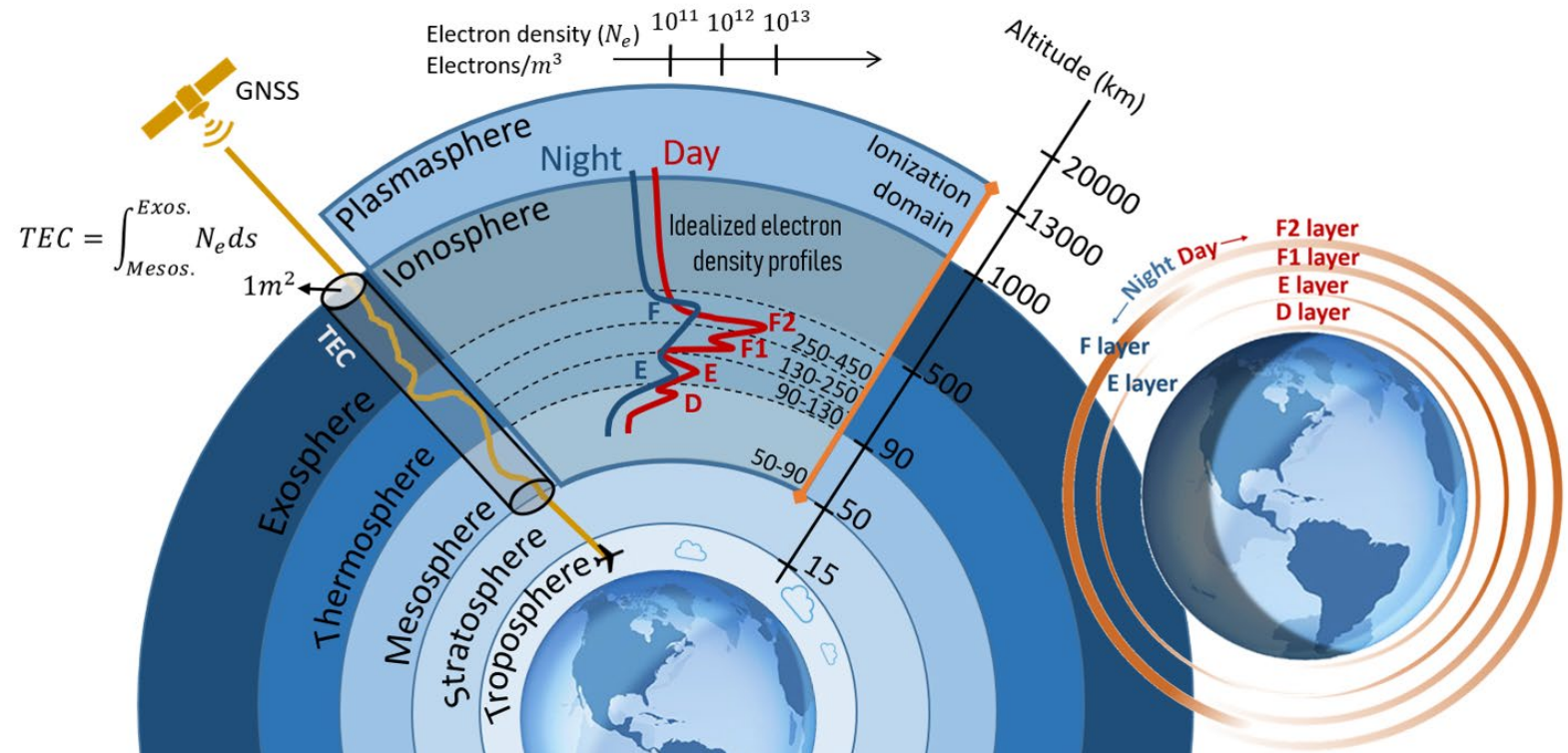
# Total Electron Content

Using the 1st order expansion, the Optical Path of a GNSS signal travelling from the satellite to a ground receiver can be expressed as:

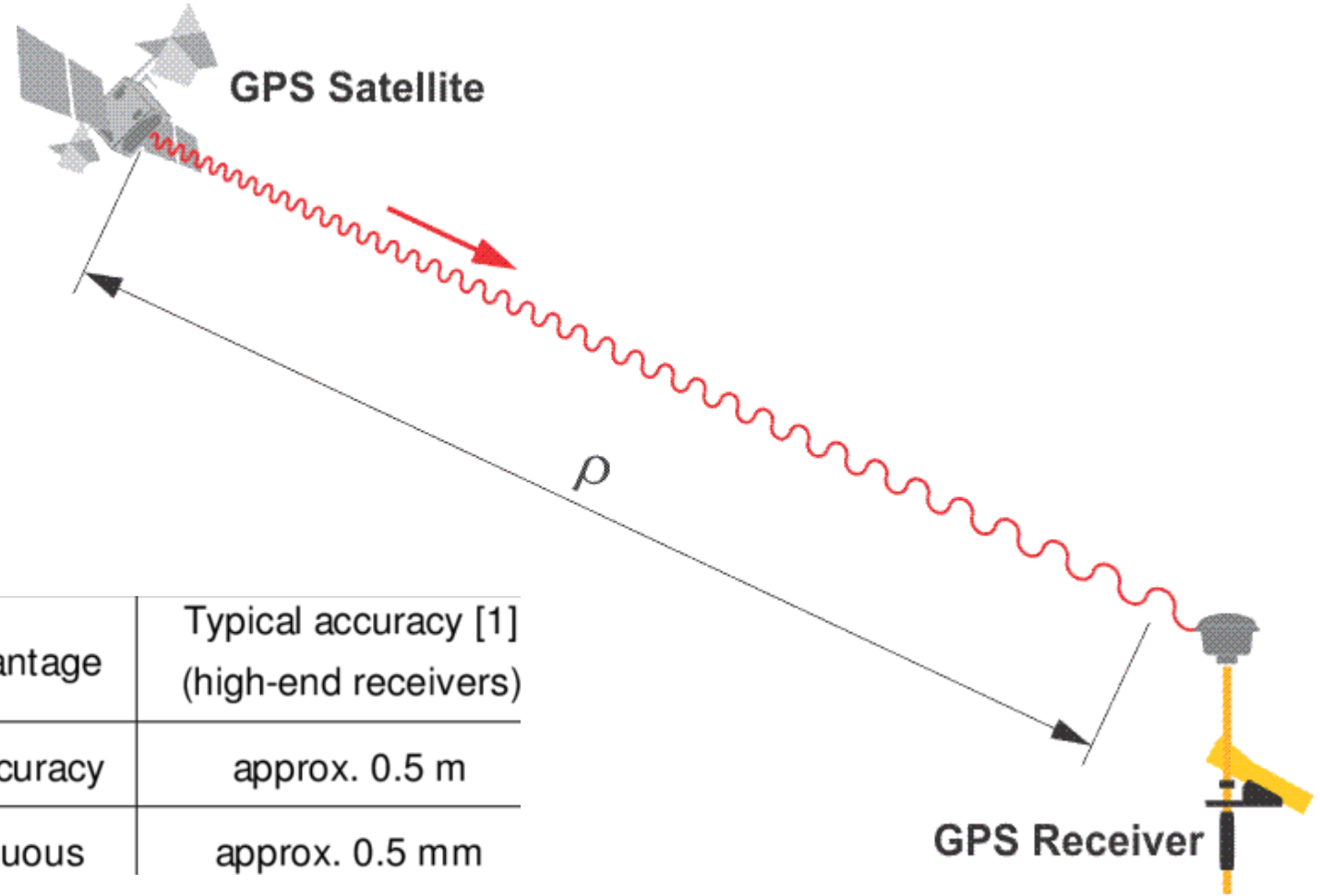
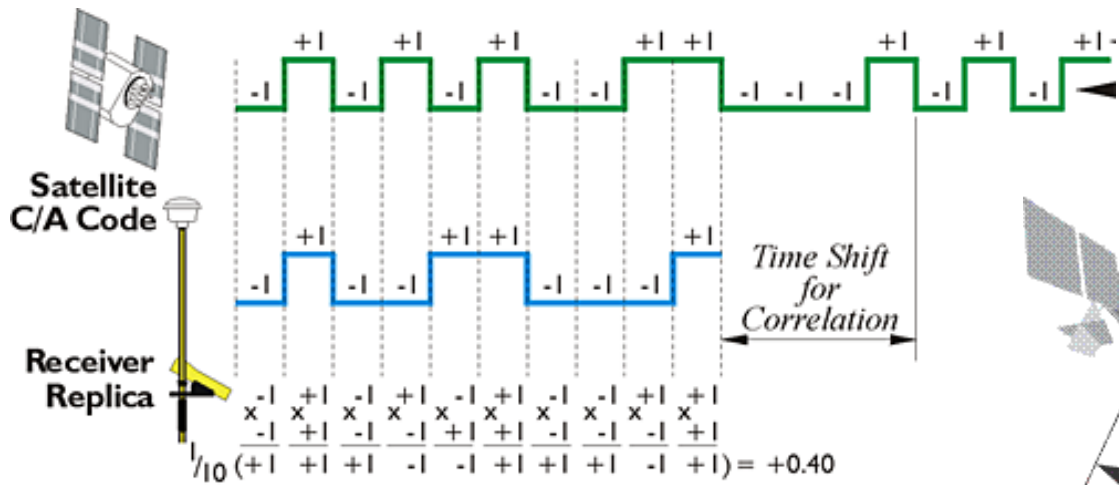
$$\Lambda = \int_{\text{satellite}}^{\text{receiver}} n ds = \int \left( 1 - \frac{40.3N_e}{f^2} \right) ds = \underbrace{S}_{\text{Geometric path}} - \frac{40.3}{f^2} \int N_e ds = S - \frac{40.3}{f^2} STEC$$

$$STEC = \int_{\text{satellite}}^{\text{receiver}} N_e ds$$

$$1 \text{ TECU} = 10^{16} \text{ e/m}^2$$



# Ionospheric Phase (L) and Code (d) Delays for GNSS



	Advantage	Disadvantage	Typical accuracy [1] (high-end receivers)
Code measurement	Unambiguous	Low accuracy	approx. 0.5 m
Carrier measurement	High accuracy	Ambiguous	approx. 0.5 mm



# Linear Combination of GNSS Measurements

## IONOSPHERE-FREE COMBINATION

$$\Phi_{\text{iono-free}} = \frac{f_1^2 \Phi_{L1} - f_2^2 \Phi_{L2}}{f_1^2 - f_2^2}$$

It removes the first order (up to 99.9%) ionospheric effect, which depends on the inverse square of the frequency



Positioning



Assessment of higher order ionospheric effects

## GEOMETRY-FREE COMBINATION

$$\Phi_{LI} = \Phi_{L1} - \Phi_{L2}$$

It cancels the geometric part of the measurement, leaving all the frequency-dependent effects (i.e., ionospheric refraction, instrumental delays, wind-up) besides multipath and measurement noise



Estimation of Total electron content

# Ionospheric Phase (L) and Code (d ) Delays for GNSS

$$L(\text{cycles}) = \frac{\Lambda}{\lambda} = \frac{1}{\lambda} \left( S - \frac{40.3}{f^2} STEC \right) = \frac{S}{\lambda} - \frac{40.3}{\lambda f^2} STEC = \frac{fS}{c} - \frac{40.3}{cf} STEC$$

$$\delta(\text{seconds}) = \frac{dL}{df} = \frac{S}{c} + \frac{40.3}{cf^2} STEC$$



$$L(\text{meters}) = L(\text{cycles}) * \lambda = S \ominus \frac{40.3}{f^2} STEC$$

$$\delta(\text{meters}) = \delta(\text{seconds}) * c = S \oplus \frac{40.3}{f^2} STEC$$

**Under the assumed approximation of the Appleton-Hartree formula:**

distance measurements using phase delay and code delay provide with an estimation of actual distance S plus a ionospheric contribution which in absolute value is the same for phase and code, but with opposite sign.

# Geometry free linear combination (ideal case)

$$L_1, P_1 = S \mp \frac{40.3 \text{ STEC}}{f_1^2}$$

$$L_2, P_2 = S \mp \frac{40.3 \text{ STEC}}{f_2^2}$$

How isolating ionospheric information? Solving the system provided by measurements at two frequencies  $f_1$  and  $f_2$  at advantage of the ionospheric investigator

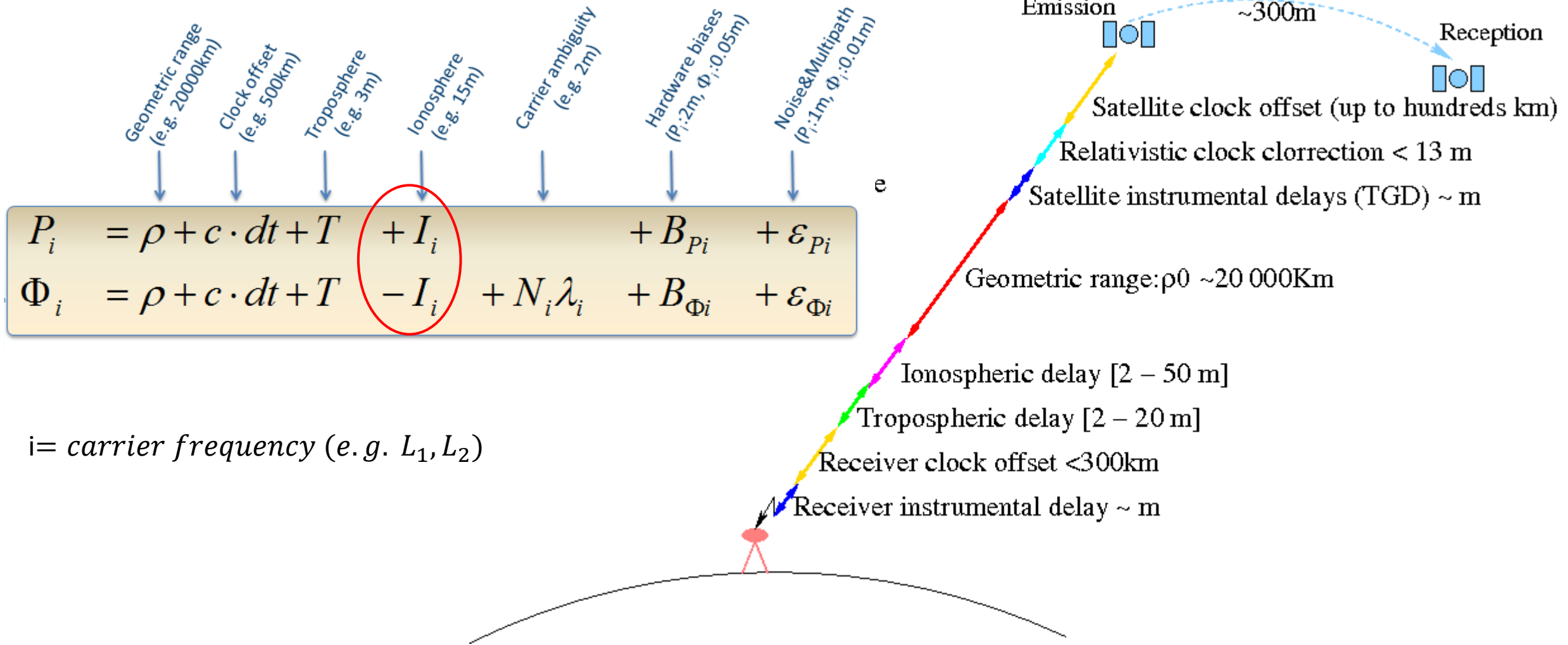
$$DPD = L_1 - L_2 = I_1 - I_2$$

$$DGD = P_2 - P_1 = I_2 - I_1$$

$$I_1 - I_2 = 40.3 \left( \frac{1}{f_2^2} - \frac{1}{f_1^2} \right) \text{ STEC}$$

But in the real world...

# Code and Carrier phase measurements equations



The ionosphere will introduce a **delay of the modulation** (the code measurement will be larger than in vacuum), and an **advance of the carrier phase** (the carrier phase measurement will be smaller than in vacuum).

# Geometry free linear combination

$$L_{\text{arc}}[m] = L_1 - L_2 = I_1 - I_2 + \cancel{T_1 - T_2} + c(\tau_{1,R} - \tau_{2,R}) + c(\tau_{1,S} - \tau_{2,S}) + \lambda_1 N_1 + \lambda_2 N_2 + \epsilon_L$$

Troposphere is assumed to be non-dispersive medium, so the tropospheric contribution is cancelled out

$$sTEC = \left[ 40.3 \left( \frac{1}{f_2^2} - \frac{1}{f_1^2} \right) \right]^{-1} * (I_1 - I_2) = k(I_1 - I_2)$$

$$L_{\text{arc}}[TECu] = sTEC + B_R + B_S + C_{\text{arc}} + \epsilon_L$$

$$B_R = c(\tau_{1,R} - \tau_{2,R})k \quad B_S = c(\tau_{1,S} - \tau_{2,S})k$$

$$C_{\text{arc}} = k(\lambda_1 N_1 + \lambda_2 N_2) \quad \epsilon_L = k\epsilon_L$$

The meaning of **Arc** in radio observations, as a series of observations carried out with *continuity* from one station to one satellite. *Continuity*: presence of satellite over the horizon of the station (astronomical arc) , no loss of lock for phase or code.

If not recoverable loss of lock occurs, two distinct arcs will be considered also if observations belong to the same "astronomical" arc.

# Geometry free linear combination (phase vs code)

Phase observables GFLC

$$L_{\text{arc}}[TECu] = s\text{TEC} + B_R + B_S + C_{\text{arc}} + \varepsilon_L$$



Code observables GFLC

$$P = s\text{TEC} + b_R + b_S + \varepsilon_P$$

Less noisy wrt code observables  
Relative measurement (ambiguity term)

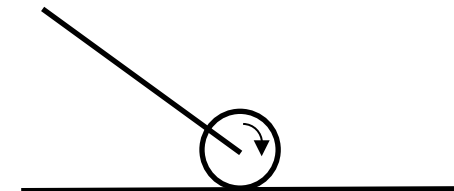
Noisier wrt phase observables  
Absolute measurement (no ambiguity term)

Phase ambiguity

Measuring phase is like measuring distance with an odometer

Apart the initial ambiguity  $\Omega$ , the user can cumulate the cycles ( $L$ ) of the incoming signal achieving very high resolution in the measurement of the distance. If some cycle is lost (**cycle slip, phase jump**), measurement re-starts with a new ambiguity

$$D = \Omega + L \cdot \lambda$$



$$\lambda = 2\pi R$$

( $R$  radius of odometer)

# Geometry free linear combination (ambiguity resolution)

Phase observables GFLC

$$L_{\text{arc}}[TECu] = s\text{TEC} + B_R + B_S + C_{\text{arc}} + \varepsilon_L$$

Code observables GFLC

$$P = s\text{TEC} + b_R + b_S + \varepsilon_P$$



$$\langle L_{\text{arc}} - P \rangle_{\text{arc}} = C_{\text{arc}} + B_R + B_S - b_R - b_S - \langle \varepsilon_P \rangle_{\text{arc}}$$

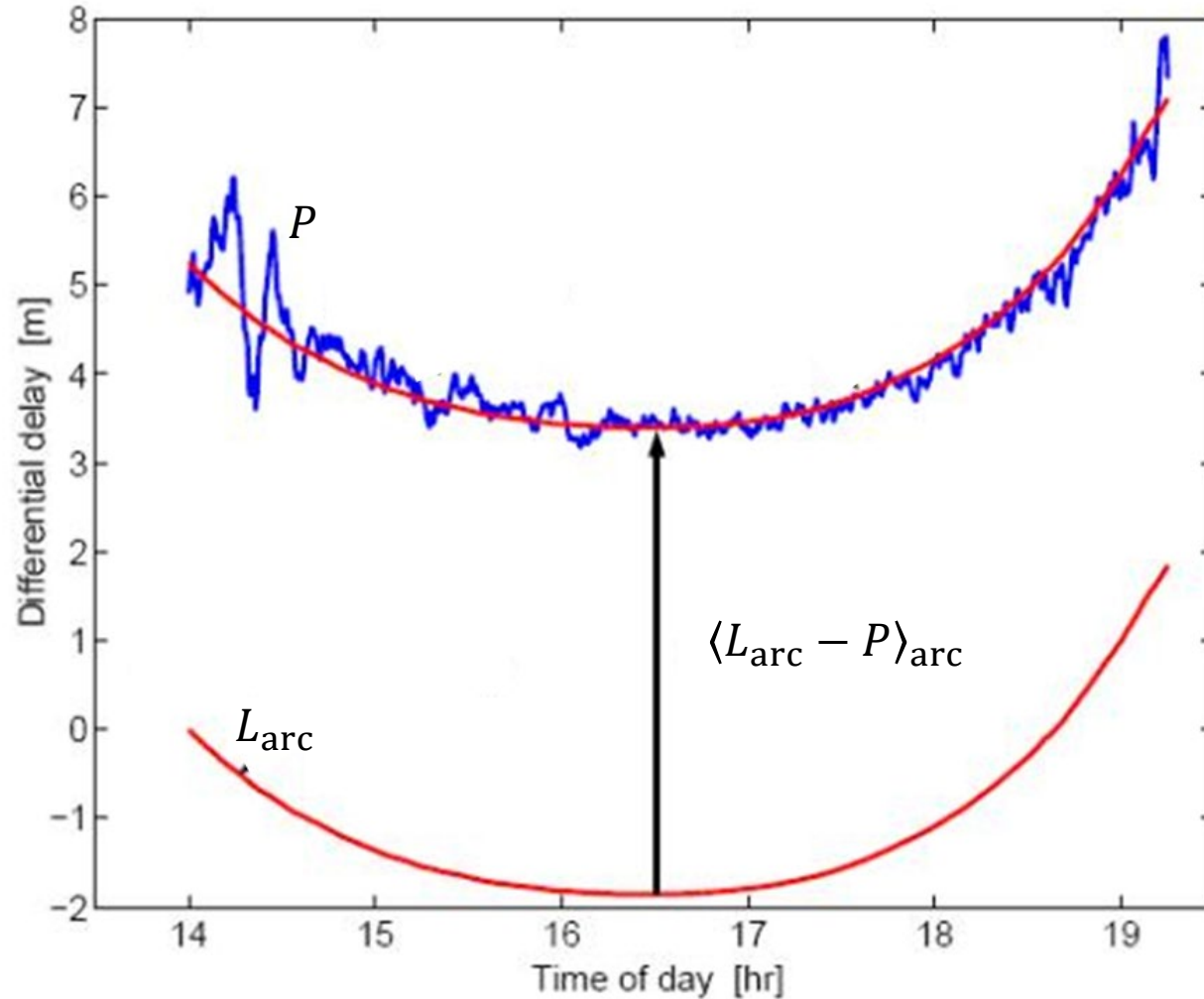
$\langle \rangle$  = mean over an arc

$$L_{\text{arc}} - \langle L_{\text{arc}} - P \rangle_{\text{arc}} = \tilde{L}_{\text{arc}} = s\text{TEC} + b_R + b_S + \langle \varepsilon_P \rangle_{\text{arc}}$$

multipath + other non-zero mean errors

# Geometry free linear combination (ambiguity resolution)

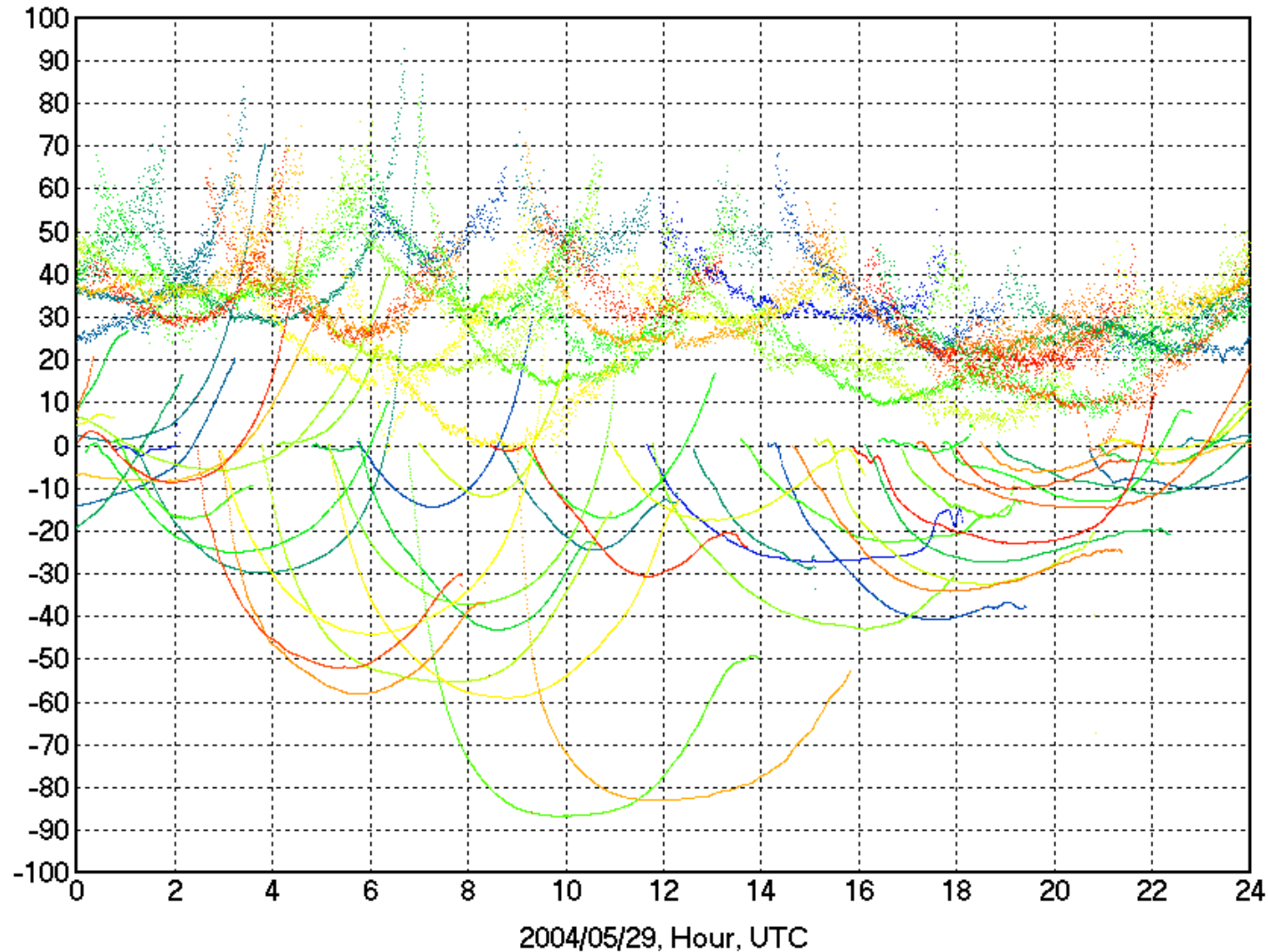
$$L_{\text{arc}} - \langle L_{\text{arc}} - P \rangle_{\text{arc}} = \tilde{L}_{\text{arc}} = \text{sTEC} + b_{\text{R}} + b_{\text{S}} + \langle \varepsilon_{\text{P}} \rangle_{\text{arc}}$$





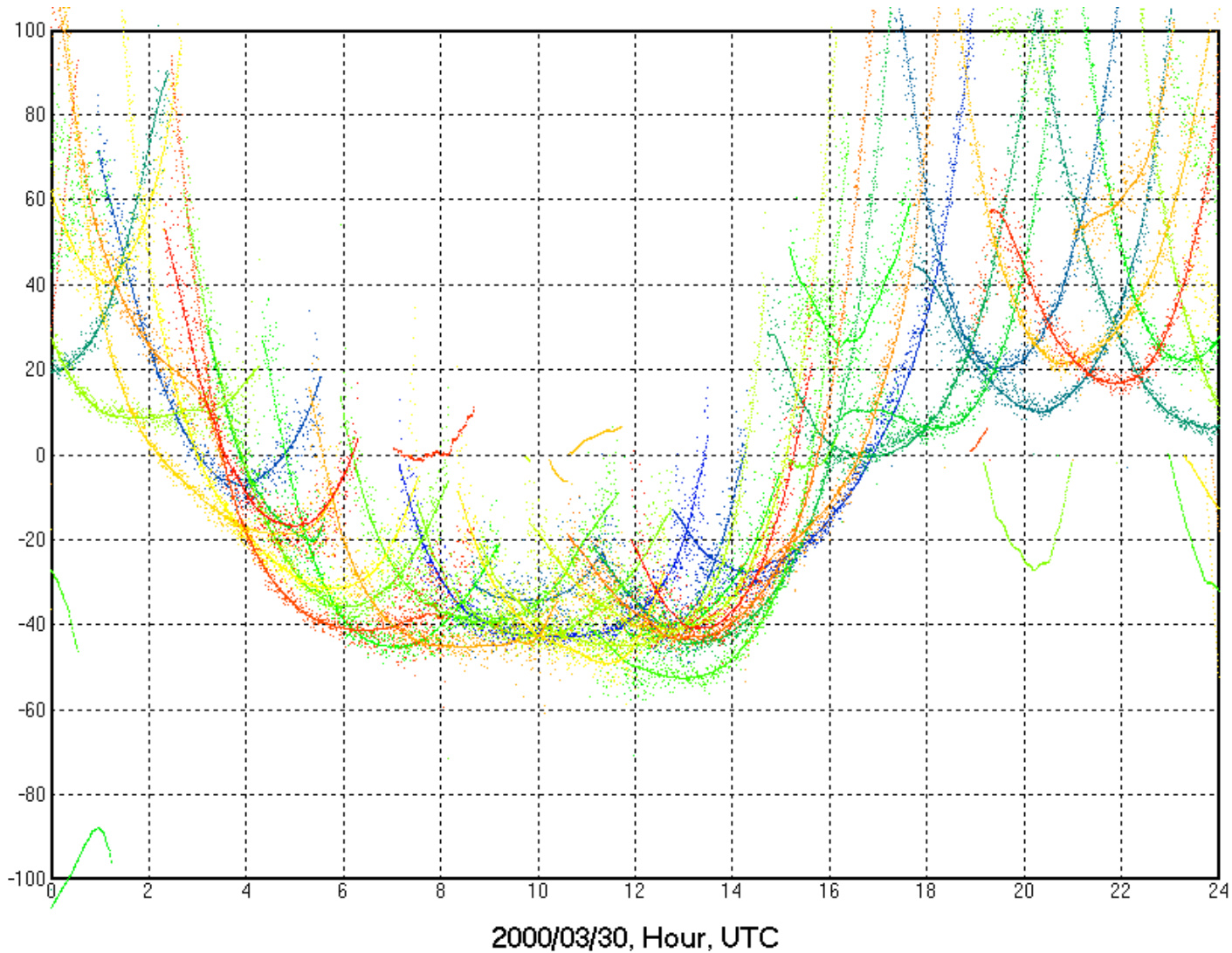
# Geometry free linear combination (ambiguity resolution)

$$L_{\text{arc}} - \langle L_{\text{arc}} - P \rangle_{\text{arc}} = \tilde{L}_{\text{arc}} = \text{sTEC} + b_{\text{R}} + b_{\text{S}} + \langle \varepsilon_{\text{P}} \rangle_{\text{arc}}$$

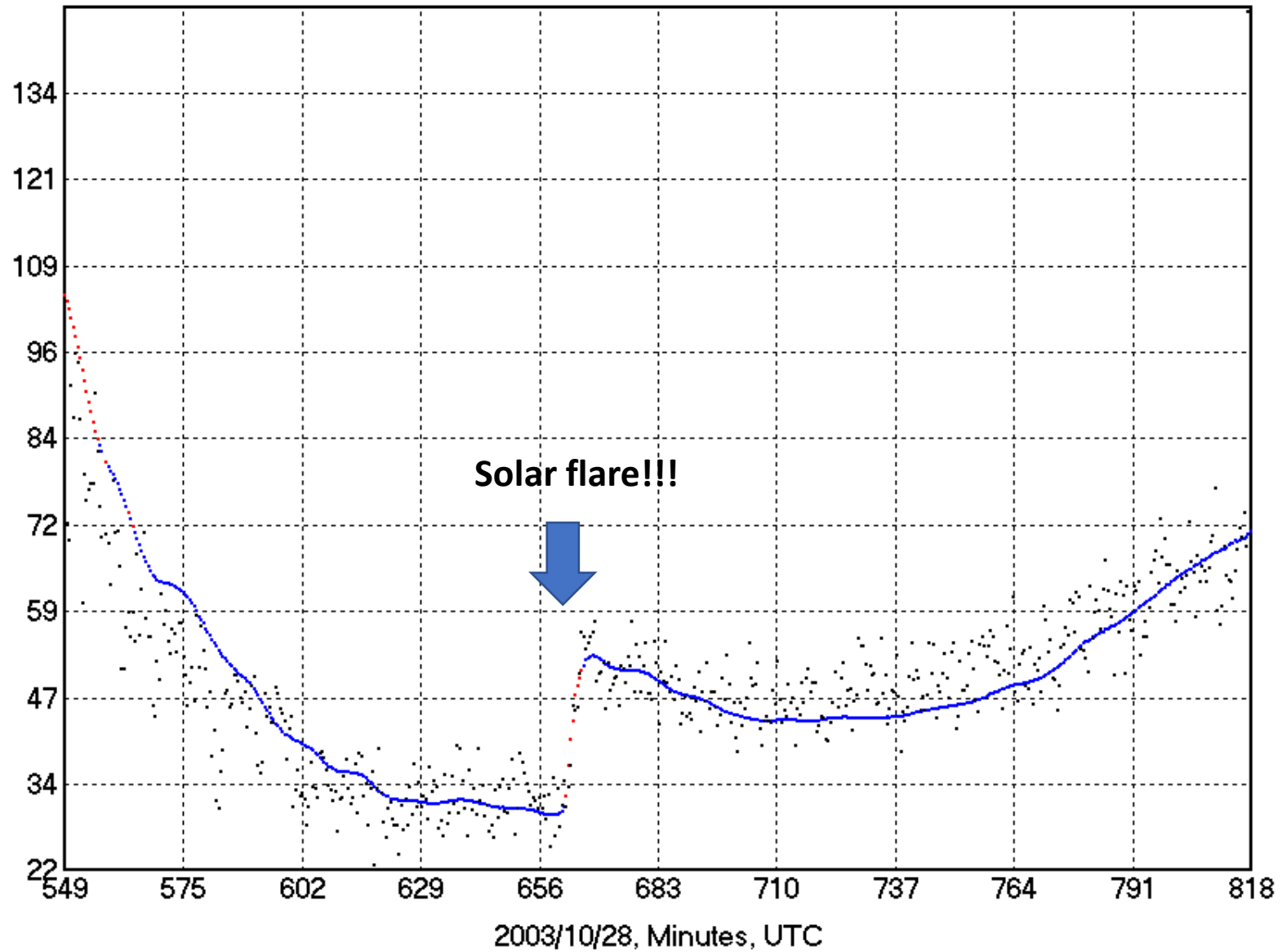


# Geometry free linear combination (ambiguity resolution)

$$L_{\text{arc}} - \langle L_{\text{arc}} - P \rangle_{\text{arc}} = \tilde{L}_{\text{arc}} = \text{sTEC} + b_{\text{R}} + b_{\text{S}} + \langle \varepsilon_{\text{P}} \rangle_{\text{arc}}$$



# Phase jump (cycle slips?)



# Calibration and Mapping

Task of the calibration

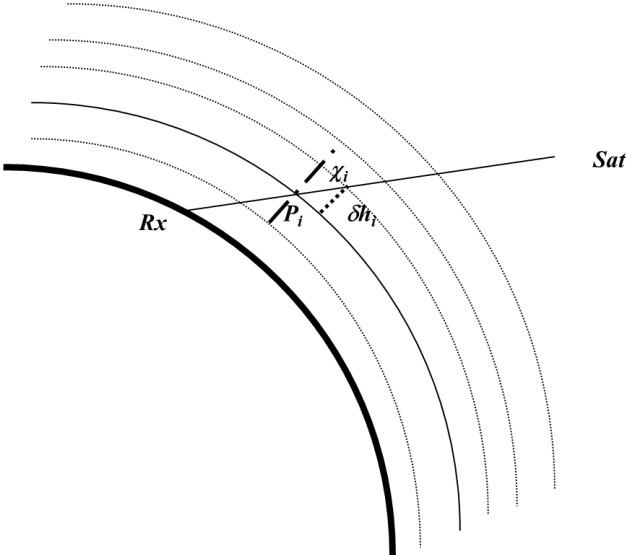
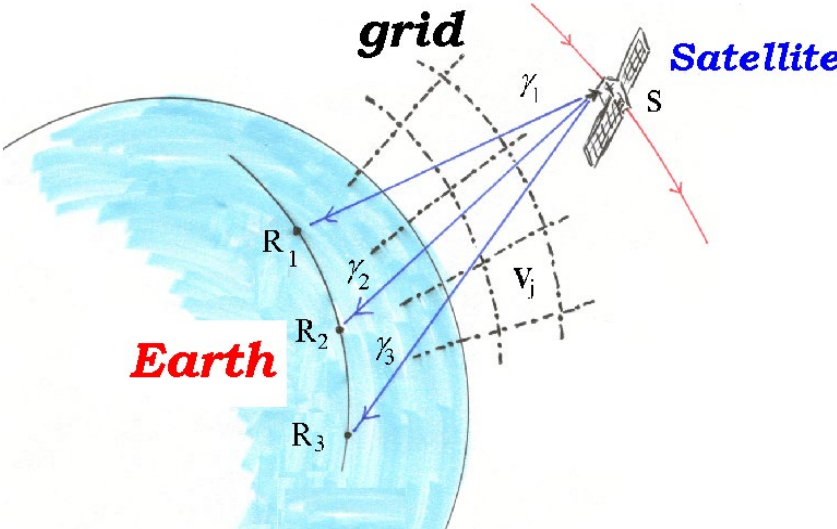
## Isolate TEC from other terms

The only way to proceed is assuming that TEC from all available observations from generic station  $R_j$  to generic satellite  $S_i$  can be expanded using proper base functions of time and position  $\Psi(P, t)$ .

This representation of TEC is the so-called **TEC Mapping**, achievable in several ways (integrated 3D electron density, or 2D shell, ...)

$$TEC(R_j, S_i, t) = \sum_k c_k \psi_k(P, t)$$

- Global many days
- Global single day
- Regional many days
- Regional single day
- Single station many days
- Single station one day



# TEC Calibration (de-biasing)

$$L_{\text{arc}} - \langle L_{\text{arc}} - P \rangle_{\text{arc}} = \tilde{L}_{\text{arc}} = \text{sTEC} + b_{\text{R}} + b_{\text{S}} + \langle \varepsilon_{\text{P}} \rangle_{\text{arc}}$$

$$\langle \varepsilon_{\text{P}} \rangle_{\text{arc}} \approx 0$$

$b_{\text{R}}, b_{\text{S}}$  evaluated independently  
every now and then

$b_{\text{R}}, b_{\text{S}}$  can be taken from **IGS  
Ionosphere Associate  
Analysis Centers (IAACs)**

$$\tilde{L}_{\text{arc}} = \text{sTEC} + \beta_{\text{arc}}$$

$$\beta_{\text{arc}} = b_{\text{R}} + b_{\text{S}} + \langle \varepsilon_{\text{P}} \rangle_{\text{arc}}$$

to be evaluated arc by arc

**Gg calibration technique** (see  
e.g. Ciruolo et al., 2007;  
Cesaroni et al., 2015; Cesaroni  
et al., 2021)

$$\sum_{j=\text{sat}} b_r^j = 0$$

$$\tilde{L}_{\text{arc}} = \tilde{L}_{\text{arc}}(\varphi, \lambda, H, t)$$

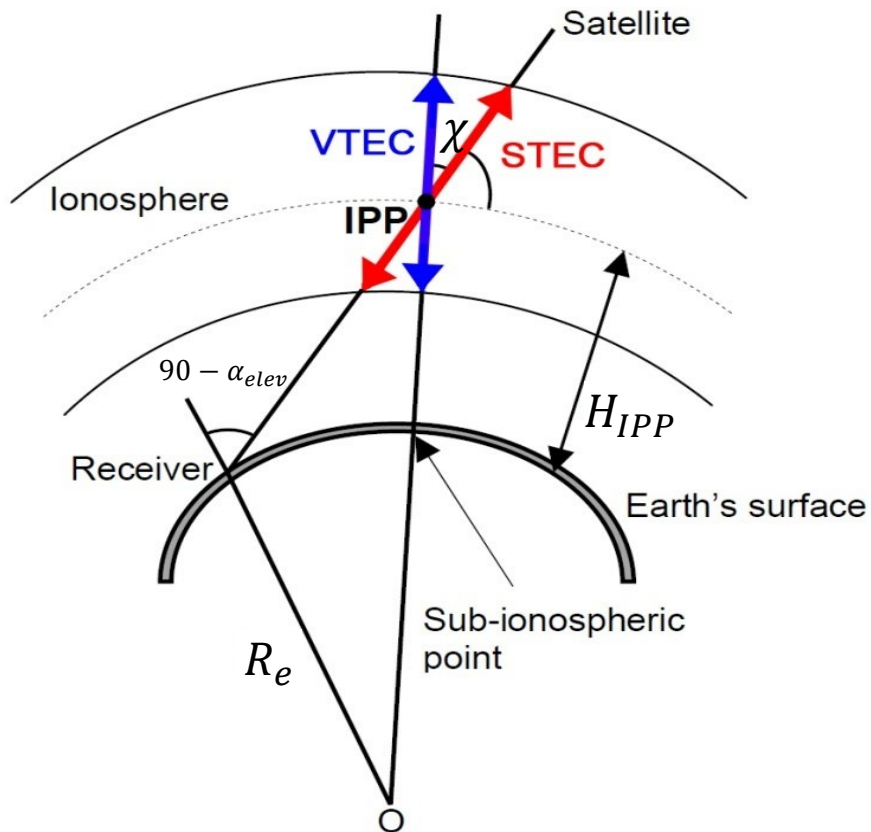
Global or regional TEC model

$b_{\text{R}}, b_{\text{S}}$  evaluated independently

# Single day – single station approach (Gg technique)

$$\tilde{L}_{arc}(R_j, S_i, t) = sTEC(R_j, S_i, t) + \beta_{arc}$$

Once selected some method of mapping has been assumed, the coefficients  $c$  of TEC expansion become a new set of unknowns to be estimated together with the “biasing” terms using standard minimization algorithms



$$\tilde{L}_{arc}(R_j, S_i, t) = \sum_k c_k \psi_k(P, t) + \beta_{arc}$$

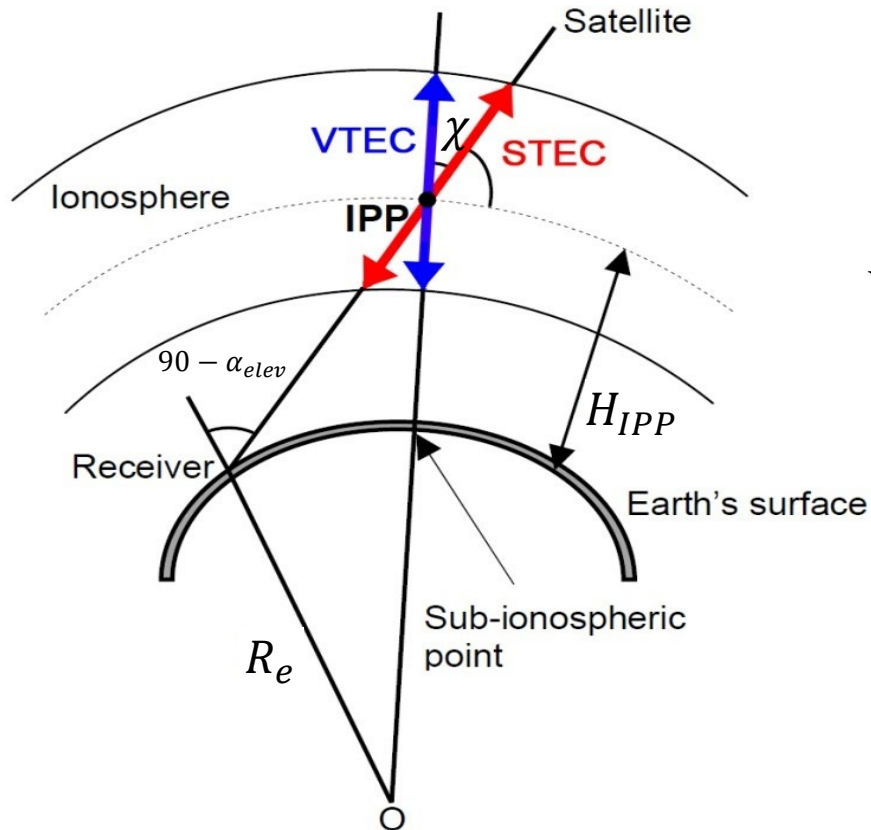
$$F(\alpha_{elev}) = \frac{1}{\sqrt{1 - \left(\frac{R_e \cos \alpha_{elev}}{R_e + H_{IPP}}\right)^2}},$$

$$STEC = VTEC \cdot \sec \chi = VTEC \cdot F(\alpha_{elev})$$

# Single day – single station approach (Gg technique)

$$sTEC = vTEC(\phi_1, \phi_2) \cdot \sec \chi$$

where  $vTEC(\phi_1, \phi_2)$  is the unknown describing a surface in the reference frame defined by a couple  $(\phi_1, \phi_2)$  over the thin shell (bi-dimensional)



$$\tilde{L}_{arc} = \sec \chi \sum_n c_n p_n(LT, Modip) + \beta_{arc}$$









vTEC is expanded as a polynomial linear in LT and of the fourth-order in Modip

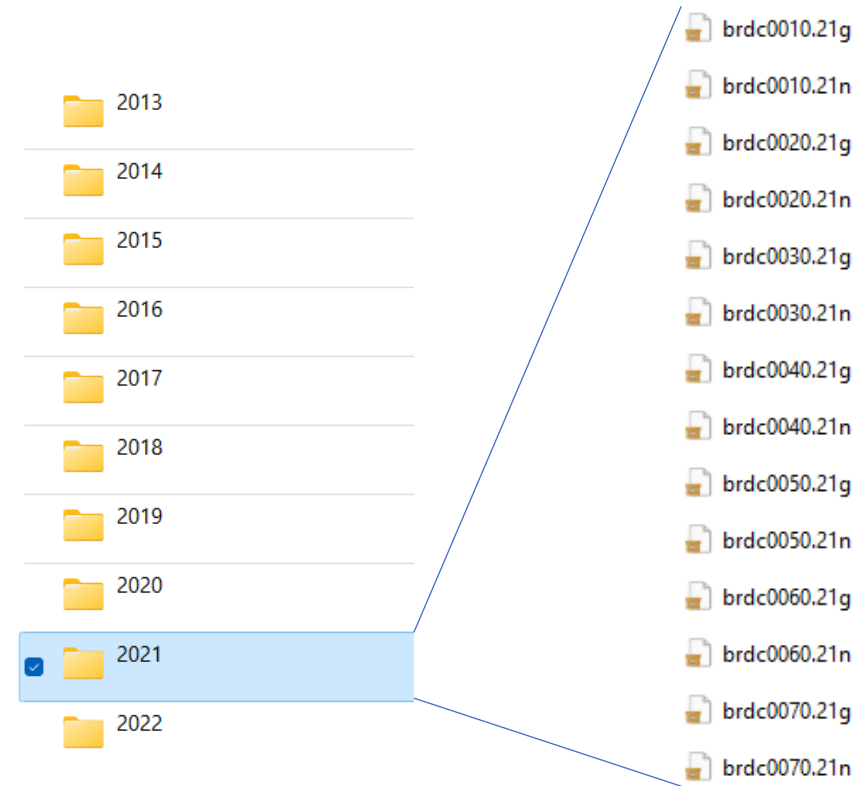
$$Modip = \arctan \left( \frac{I}{\sqrt{\cos \varphi}} \right)$$

Magnetic dip at 350 km

Geographic latitude

# Gg software – overview

 ingr	Station(s) directory(ies)
 OutPut	Output directory
 Calibration_settings.txt	File containing the calibration settings (see next slides)
 crx2rmx.exe	Hatanaka decompression software (not mandatory)
 GNSS_calibration.exe	Executable file
 LonLats.dip	Modip lookup table
 prepare_RINEX_for_cal.m	MATLAB script to organize input files (not mandatory)
 read_ciraolo_output_fast_STEC.m	MATLAB script to read output files (not mandatory)



Gg calibration software is a windows package capable of evaluating single-station calibrated TEC from daily RINEX (v2.11) observational (ssssdoy0.yyo) and navigational (brdcdoy0.yyn/g) files from GPS and GLONASS satellites.

A new (Python) version of the software capable of processing GALILEO satellites will be available in the next future



# Gg software – RINEX obs files (v 2.11)

2.11	OBSERVATION DATA	G (GPS)					RINEX VERSION / TYPE	RINEX type/version
teqc	2016Apr1	20220103 10:03:19UTC	PGM / RUN BY / DATE					
Linux	2.4.21-27.ELsmp Opteron gcc Linux	x86_64 =+	COMMENT					Comments (ignored by the software)
teqc	2019Feb25	20220103 10:03:12UTC	COMMENT					
Linux	2.6.32-573.12.1.x86_64 x86_64 gcc	-static Linux 64 =+	COMMENT					
BIT 2 OF LLI FLAGS DATA COLLECTED UNDER A/S CONDITION							COMMENT	
INGR			MARKER NAME					Station ID
Giuseppe_Casula	INGV	OBSERVER / AGENCY						
495176	LEICA GRX1200+GNSS	9.20/6.404	REC # / TYPE / VERS					
103144	LEIAT504	NONE	ANT # / TYPE					
4646743.0089	1031406.1366	4231452.7511	APPROX POSITION XYZ					Position of the receiver in ECEF coordinates (WGS84)
0.0000	0.0000	0.0000	ANTENNA: DELTA H/E/N					
1	1	WAVELENGTH FACT L1/2						
6	L1	L2	C1	P2	S1	S2	# / TYPES OF OBSERV	Number and type of GNSS observables available in the file
30.0000							INTERVAL	Sampling time of the observables
Forced Modulo Decimation to 30 seconds							COMMENT	
SPIDER							COMMENT	
Made by Spider v4.0.1							COMMENT	
Project creator:							COMMENT	
DAnastasio-Selvaggi							COMMENT	
SNR is mapped to RINEX snr flag value [0-9]							COMMENT	
L1 & L2: min(max(int(snr dBHz/6), 0), 9)							COMMENT	
2022	1	1	0	0	0.0000000	GPS	TIME OF FIRST OBS	Time of the first observables
							END OF HEADER	End of header line
22	1	1	0	0	0.0000000	0 8G32G23G21G08G10G16G01G27		YY MM DD HH MM SS.sss #sat in view at that particular epoch PRN
125160945.675	7	97528002.27147	23817336.360	23817334.140	43.400			
43.450								
120474269.007	7	93876044.88846	22925487.360	22925484.220	44.250			
41.650								
117805562.767	7	91796539.74646	22417650.580	22417646.040	45.800			
41.600								
109799902.880	8	85558367.14648	20894226.300	20894224.900	49.700			
49.700								
110595504.787	8	86178319.11648	21045624.060	21045622.460	50.800			
50.800								
120303830.786	7	93743230.52946	22893058.300	22893054.420	44.850			
40.200								
128382932.311	5	100038654.89545	24430460.160	24430458.120	35.000			
35.100								
106285891.940	8	82820170.27248	20225530.200	20225528.200	50.900			
50.950								
22	1	1	0	0	30.0000000	0 8G32G23G21G08G10G16G01G27		Observations block
125076168.884	6	97461942.45247	23801203.460	23801201.880	41.950			
42.000								
120564314.405	7	93946210.12546	22942622.380	22942619.080	44.050			

# Gg software – RINEX nav files (v 2.11)

```

2.11 N: GPS NAV DATA RINEX VERSION / TYPE
teqc 2019Feb25 BKG Frankfurt 20220105 09:17:40UTCPCGM / RUN BY / DATE
Linux 2.6.32-573.12.1.x86_64|x86_64|gcc -static|Linux 64|+= COMMENT
Converto v3.5.6 IGN 20220105 000521 UTC COMMENT
Linux 2.6.32-573.12.1.x86_64|x86_64|gcc|Linux 64|+= COMMENT
Concatenated RINEX files (31) COMMENT
    
```

RINEX type/version

```

END OF HEADER
    
```

Date and time

```

1 22 1 4 0 0 0.0 4.665437154472D-04-1.000444171950D-11 0.000000000000D+00
7.300000000000D+01-4.275000000000D+01 4.347323940635D-09-5.174246376131D-01
-2.102926373482D-06 1.122187799774D-02-4.135072231293D-07 5.153668924332D+03
1.728000000000D+05 5.587935447693D-09-1.159150957942D+00 1.192092895508D-07
9.863567403002D-01 3.972812500000D+02 8.845144020666D-01-8.498925442930D-09
-1.964367538078D-10 1.000000000000D+00 2.191000000000D+03 0.000000000000D+00
2.000000000000D+00 0.000000000000D+00 5.122274160385D-09 7.300000000000D+01
1.656180000000D+05 4.000000000000D+00
    
```

Orbital parameters

```

1 22 1 4 2 0 0.0 4.664720036089D-04-1.000444171950D-11 0.000000000000D+00
7.400000000000D+01-2.246875000000D+01 4.259105980288D-09 5.327547024575D-01
-1.041218638420D-06 1.122482249048D-02-1.564621925354D-07 5.153674575806D+03
1.800000000000D+05-3.725290298462D-08-1.159212006976D+00 1.117587089539D-07
9.863553037147D-01 3.930312500000D+02 8.845155153473D-01-8.485710606765D-09
-2.196520065306D-10 1.000000000000D+00 2.191000000000D+03 0.000000000000D+00
2.000000000000D+00 0.000000000000D+00 5.122274160385D-09 7.400000000000D+01
1.728180000000D+05 4.000000000000D+00
1 22 1 4 4 0 0.0 4.664002917707D-04-9.890754881781D-12 0.000000000000D+00
7.500000000000D+01-6.875000000000D+00 4.368396246952D-09 1.583143868868D+00
-5.085021257401D-07 1.122385857161D-02-4.153698682785D-07 5.153669784546D+03
1.872000000000D+05-2.533197402954D-07-1.159273638252D+00 7.264316082001D-08
9.863543220967D-01 4.028437500000D+02 8.843068183799D-01-8.386063598924D-09
-8.857511808063D-11 1.000000000000D+00
2.000000000000D+00 0.000000000000D+00
1.800180000000D+05 4.000000000000D+00
1 22 1 4 22 0 0.0 4.657572135329D-04-
1.700000000000D+01 6.968750000000D+00
5.103647708893D-07 1.122514088638D-02-
2.520000000000D+05 2.402812242508D-07-
9.863576955857D-01 3.972812500000D+02
3.571577341961D-13 1.000000000000D+00
2.000000000000D+00 0.000000000000D+00
2.471880000000D+05 4.000000000000D+00
    
```

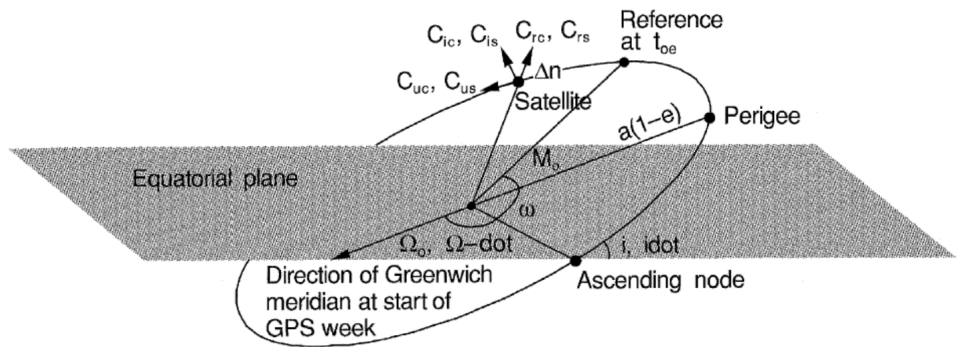


TABLE A4 GPS NAVIGATION MESSAGE FILE - DATA RECORD DESCRIPTION		
OBS. RECORD	DESCRIPTION	FORMAT
PRN / EPOCH / SV CLK	- Satellite PRN number - Epoch: Toc - Time of Clock year (2 digits, padded with 0 if necessary) month day hour minute second	I2,           1X,I2.2, 1X,I2, 1X,I2, 1X,I2, 1X,I2, F5.1, 3D19.12
	- SV clock bias (seconds) - SV clock drift (sec/sec) - SV clock drift rate (sec/sec2)	   *)
BROADCAST ORBIT - 1	- IODE Issue of Data, Ephemeris - Crs (meters) - Delta n (radians/sec) - M0 (radians)	3X,4D19.12
BROADCAST ORBIT - 2	- Cuc (radians) - e Eccentricity - Cus (radians) - sqrt(A) (sqrt(m))	3X,4D19.12
BROADCAST ORBIT - 3	- Toe Time of Ephemeris (sec of GPS week) - Cic (radians) - OMEGA (radians) - CIS (radians)	3X,4D19.12
BROADCAST ORBIT - 4	- i0 (radians) - Crc (meters) - omega (radians) - OMEGA DOT (radians/sec)	3X,4D19.12
BROADCAST ORBIT - 5	- IDOT (radians/sec) - Codes on L2 channel - GPS Week # (to go with TOE) - Continuous number, not mod(1024)! - L2 P data flag	3X,4D19.12
BROADCAST ORBIT - 6	- SV accuracy (meters) - SV health (bits 17-22 w 3 sf 1) - TGD (seconds) - IODC Issue of Data, Clock	3X,4D19.12
BROADCAST ORBIT - 7	- Transmission time of message (sec of GPS week, derived e.g. from Z-count in Hand Over Word (HOW)) - Fit interval (hours) (see ICD-GPS-200, 20.3.4.4) Zero if not known - spare - spare	3X,4D19.12

# Gg software – Calibration settings file

Output Sampling Time, Minutes	0	Out Sampling time (0 for the same resolution of the input Rinex files)
Output Minimum Elevation, Deg	20	Elevation mask (data «below» this value will not appear in the output files)
Solution: Arcs(A), Hardware biases (B)	A	Not implemented in this version (maintain «A»)
Select Output Format	3	Format of the output files (3 if you want to use the MATLAB scripts)
Folder of BRDC files	C:\...	Folder in which you put the BRDC files
Disable Rejection of First and Last Day	False	Maintain «FALSE» for reliable solutions
Shell Height	350	Height of the ionospheric thin shell in km
Discard GLONASS	False	Set «TRUE» if you want to ignore GLONASS observations

# Gg software – Output files

00000	Z00	000.000	90.000	-063.243	+17.621	+005.29	+005.29
00000	G01	034.739	72.035	-062.664	+18.307	+004.05	+003.87
00000	G03	140.776	27.940	-059.881	+13.480	+018.04	+009.88
00000	G07	233.164	63.521	-064.475	+16.627	+007.58	+006.87
00000	G14	326.331	25.006	-066.710	+22.305	+004.76	+002.43
00000	G17	283.745	26.685	-068.815	+18.727	+007.34	+003.90
00000	G21	034.265	40.255	-061.197	+20.322	+005.54	+003.82
00000	G30	288.242	51.517	-065.577	+18.232	+005.27	+004.25
00000	R19	006.021	38.160	-062.832	+21.151	+003.33	+002.21
00000	R20	269.494	51.845	-065.661	+17.480	+006.16	+004.99
00015	Z00	000.000	90.000	-063.243	+17.621	+005.31	+005.31
00015	G01	034.565	71.913	-062.662	+18.315	+004.04	+003.86
00015	G03	140.674	28.013	-059.882	+13.496	+017.88	+009.80
00015	G07	232.913	63.447	-064.475	+16.619	+007.59	+006.88
00015	G14	326.395	25.088	-066.694	+22.294	+004.75	+002.43
00015	G17	283.870	26.714	-068.807	+18.737	+007.32	+003.89
00015	G21	034.242	40.151	-061.191	+20.333	+005.54	+003.81
00015	G30	288.057	51.561	-065.575	+18.224	+005.27	+004.26
00015	R19	006.099	38.045	-062.825	+21.165	+003.31	+002.20
00015	R20	269.739	51.863	-065.660	+17.490	+006.16	+004.99
00030	Z00	000.000	90.000	-063.243	+17.621	+005.32	+005.32
00030	G01	034.393	71.790	-062.661	+18.322	+004.03	+003.85
00030	G03	140.571	28.085	-059.883	+13.513	+017.74	+009.74
00030	G07	232.664	63.372	-064.474	+16.611	+007.61	+006.89
00030	G14	326.458	25.170	-066.677	+22.283	+004.75	+002.44
00030	G17	283.996	26.743	-068.798	+18.748	+007.33	+003.90
00030	G21	034.220	40.048	-061.185	+20.343	+005.54	+003.81
00030	G30	287.871	51.605	-065.574	+18.216	+005.28	+004.27
00030	R19	006.176	37.929	-062.818	+21.178	+003.38	+002.24
00030	R20	269.986	51.881	-065.658	+17.500	+006.16	+004.99

Seconds since the beginning of the day

Satellite (G for GPS, R for GLONASS)

Azimuth

Elevation

Longitude of the IPP

Latitude of the IPP

STEC

VTEC

# Gg software

**Let's go for a live session!!!!**

[https://www.dropbox.com/sh/5ti51wl52e977ns/AACxxaCm8eBrveQHTW4xyFM\\_a?dl=0](https://www.dropbox.com/sh/5ti51wl52e977ns/AACxxaCm8eBrveQHTW4xyFM_a?dl=0)