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WORKING PARTY ON "FRACTURE PHYSICS"
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CRACK DISLOCATIONS

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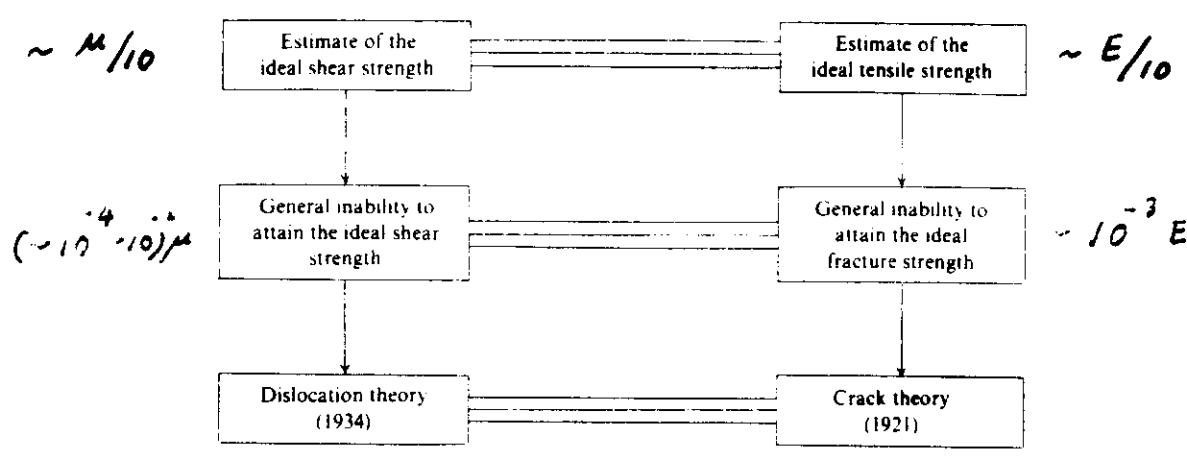
These are preliminary lecture notes, intended only for distribution to participants.

DISLOCATIONS AND CRACKS

Deformation
(non- elastically)

Fracture

Table I
 The foundations of dislocation and crack theories



A particularly important example of a material with a low fracture strength is ordinary window glass which breaks at a stress of about 10^4 psi as compared with an ideal tensile strength of about 10^6 psi. The reason is that its surface contains cracks, typically about 10^{-4} cm deep, which arise for example during abrasion: these cracks can be detected [12] by decorating the surface with sodium vapor. Near-ideal tensile strengths have been obtained for some bulk solids such as glass, silica and sapphire, by taking exceptional care to avoid surface defects: thus, for example, a tensile strength of $E/20$ has been measured for 0.25 in (6 mm) diameter glass rods [13]. In this context it should be noted that fibres and whiskers have tensile strengths near the ideal value because their mode of preparation and handling is conducive to a good surface finish [14]. The importance of surface stress concentrations is recognized in many practical situations: thus the surface of glass in motor car windscreens is

- Crack: 1. as a distribution of virtual "crack" dislocations
- 2. Dislocation shielding and anti-shielding.
- (3. Dislocation role on crack nucleation and growth.)

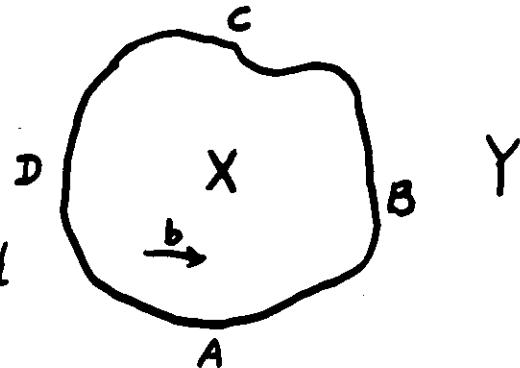
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DISLOCATIONS

Dislocation:

The boundary line between a slipped and an unslipped area is called a dislocation line (ABCD).

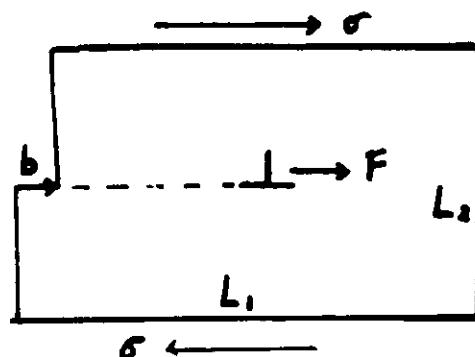


A dislocation line has never end within a crystal; it may form a closed ring, or can be a free surface, or be joined to other dislocation lines.

Imperfections in real crystal — The observed yield strengths of the softest crystals are in the range $10^{-5}\mu$ to $10^{-4}\mu$. i.e. several orders of magnitude below the shear strength of a perfect lattice.

Force on a dislocation:

$$F = (\sigma L_1 b) / L_1 \\ \approx \sigma b$$

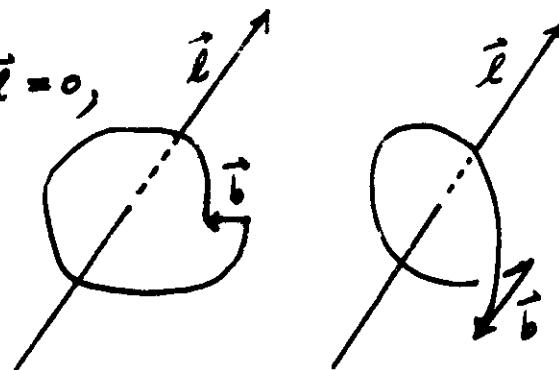


Screw and Edge Dislocations

Edge: $\vec{b}_e \times \vec{e} \neq 0, \vec{b}_e \cdot \vec{e} = 0$

Screw: $\vec{b}_s \times \vec{e} = 0$
 $\vec{b}_s \cdot \vec{e} \neq 0$

Mixed type: $\vec{b} = \vec{b}_s + \vec{b}_e$



Elastic field of dislocations

an edge:

$$\sigma_{rr} = \sigma_{\theta\theta} = -D \sin \theta/r, \quad \sigma_{r\theta} = D \cos \theta/r$$

$$\text{where } D = \mu b / 2\pi(1-\nu) \quad (= A)$$

an screw

$$\sigma_{\theta\theta} (= \sigma_{\phi\phi}) = \frac{\mu b}{2\pi r}$$

$1 - D$:

$$\sigma^2 = \frac{A}{x-x'} \quad (= \frac{A}{r})$$



An infinite isotropic continuum:

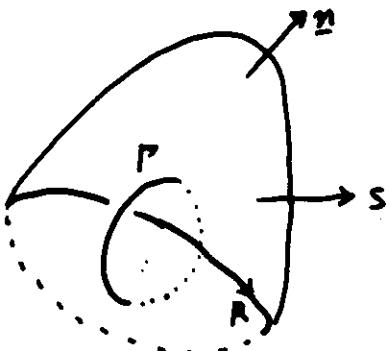
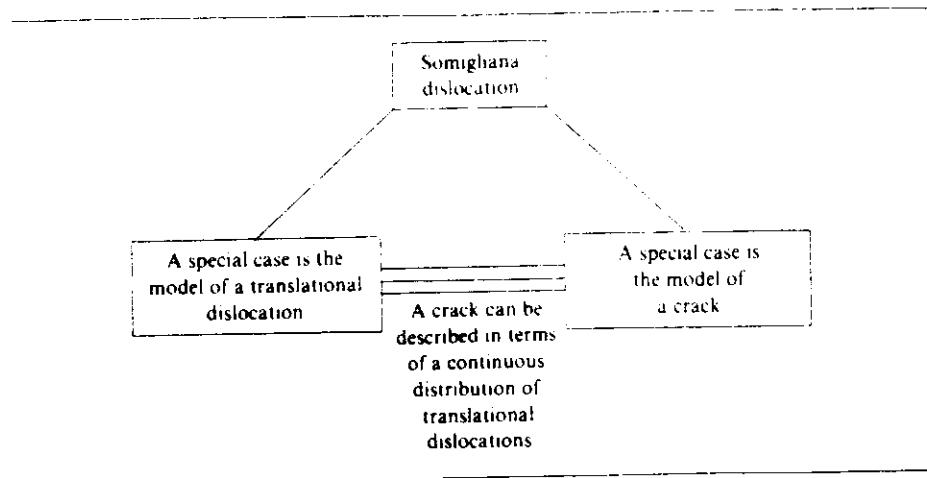


Table 2
Schematic representation of the relationship between the elastic continuum models of a translational dislocation and a crack



A cut is made over an open surface S with unit outward normal n ; the positive side of the cut to which n points is displaced with respect to the negative side by a small constant vector d , which represents the magnitude and direction of the slip displacement; material is removed, gaps are filled, the solid being rewelded.

The translational dislocation ring R is the boundary of S .

$$-d = b \quad \text{Burgers vector (RH/FS convention)}$$

Crack — not rewelded.

Crack dislocations:

1. Inserting material. with $D_i(x_i) > 0$ ($x_i > 0$); $D_i(x_i) < 0$ ($x_i < 0$)
2. Generating stress distribution. $p_{i,i}^A(x_i)$ along $x_i = 0$,
3. An additional stress $p_{i,i}^D$; $D_i(x_i)$ is adjusted so that $p_{i,i}^A + p_{i,i}^D = 0$
4. The inserted material being removed without disturbing stress field.

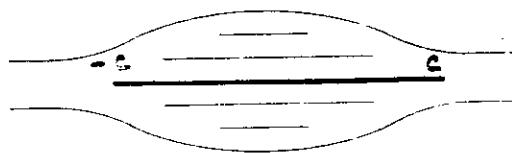


Fig. 10 The model used [28] for the interpretation of crack dislocations: material initially inserted

Then, in the region: traction-free — the behaviour of a crack

These are crack dislocations

Crack dislocation

virtual

cuts unwelded

$$p_i \cdot n_j = 0 \text{ on } S$$

$$\sigma^A = 0; D_i(x_i) = 0$$

ordinary dislocation

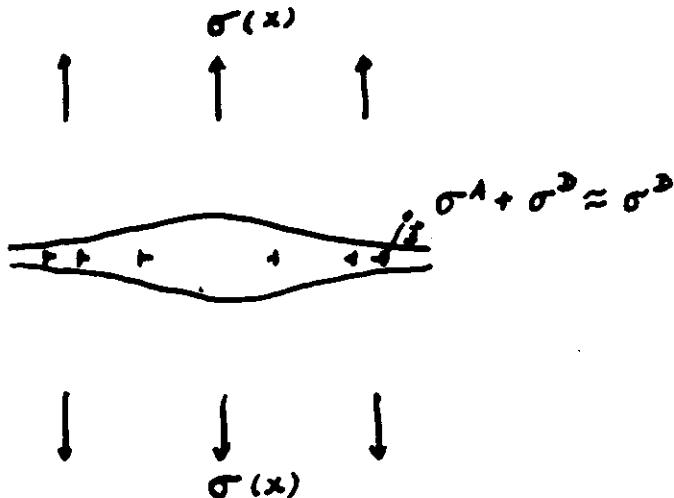
real

cuts rewelded

$p_i \cdot n_j$ are continuous
across S

$$\sigma^A = \sigma; D_i(x_i) \neq 0$$

Crack as a continuous distribution of dislocations



Traction-free surface condition (in crack region)

$$\sigma^1 + \sigma^2 = 0$$

$$\sigma_{yy}^1(x) + A \int_{-a}^a \frac{D(x') dx'}{x - x'} = 0, \quad A = \frac{Gb}{2\pi(1-\nu)}$$

$$\sigma_{ij}^2(x) = b \int_{-a}^a D_y(x') \sigma_{ij}^{yy}(x-x'; y, z) dx'$$

$$\sigma^2(s) \approx \frac{K^2}{(2\pi)^{\frac{1}{2}}} s^{-\frac{1}{2}} \quad (x = s + a)$$

$$D(s) \approx \frac{K^2}{\pi A (2\pi)^{\frac{1}{2}}} (-s)^{-\frac{1}{2}}$$

$$K^2 = (2\pi)^{\frac{1}{2}} \lim_{s \rightarrow 0} s^{\frac{1}{2}} \sigma^2(s)$$

$$(K^2)^2 = 2\pi^2 A^2 \lim_{x \rightarrow a} (a-x) D^2(x)$$

Bilby and Eshelby: 1968.
in "Fracture"

The solution of the Integral Equation

$$D(x) = \frac{1}{A\pi} \int_{-a}^a \left(\frac{a^2 - x'^2}{a^2 - x^2} \right)^{1/2} \frac{\sigma^A(x') dx'}{x - x'} + \frac{n}{\pi(a^2 - x^2)^{1/2}}$$

where, $n = \int_{-a}^a D(x) dx$

= the total number of dislocations

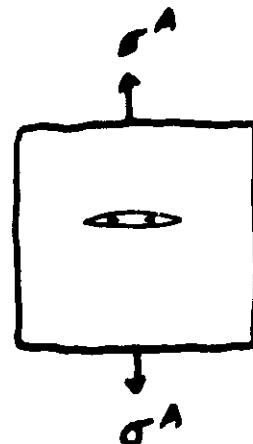
It may be determined by boundary condition.

Central Crack under Uniform Tension

$$\sigma^A = P \text{ (const.)}$$

$$n = 0$$

$$D(x) = \frac{Px}{\pi A} (a^2 - x^2)^{1/2}$$

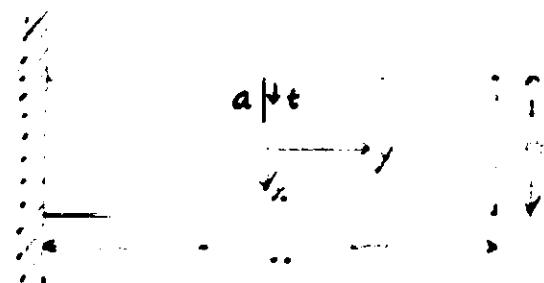


$$K = P \sqrt{\pi a}$$

The same as obtained in fracture mechanics.

Graphical Solution

$$\sigma_y(x) \Big|_{y=0} = -\alpha x$$



where,

$$\alpha = \frac{M}{2I}, \quad I = \frac{1}{12} B (W-a)^3 \quad \left(\frac{W+a}{2}, 0 \right)$$

$$\text{and } x = t - \frac{W-a}{2} - a = t - \frac{W+a}{2}$$

The integral equation becomes

$$-\alpha \left(t - \frac{W+a}{2} \right) + \int_0^a \frac{D(t') dt'}{t - t'} = 0$$

The solution is (under the boundary condition

$$t=0, \quad D(t) = 0$$

$$D(t) = \frac{\alpha}{4\pi} \left[(a^2 - t^2)^{\frac{1}{2}} - \frac{a^2}{(a^2 - t^2)^{\frac{1}{2}}} \right] + \frac{\alpha(W+a)t}{4\pi A(a^2 - t^2)^{\frac{1}{2}}}$$

$$\text{Then, } K = \frac{MY(a/w)}{B(W-a)^{3/2}}$$

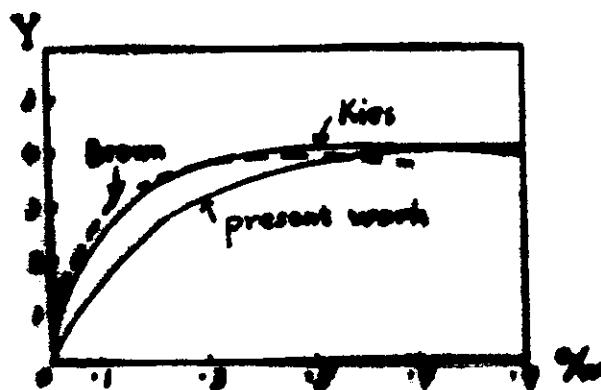
$$Y(\frac{a}{w}) = 3\sqrt{\pi} \left(\frac{a}{w} \right)^{1/2} \left(1 - \frac{a}{w} \right)^{-1/2}$$

This function may be improved by adding a modified term:

$$Y\left(\frac{a}{w}\right) = 3\sqrt{\pi} \left(\frac{a}{w}\right)^{\frac{1}{2}} \left[1 - \frac{a}{w} + \beta \left(\frac{a}{w}\right)^2 \right]^{-\frac{1}{2}}$$

β may be determined by Comparing to Wilson's analysis $Y_2\left(\frac{a}{w} = 1\right) = 4$, $\beta_2 = 1.74$

$Y \backslash \frac{a}{w}$	0	0.1	0.3	0.5	0.7	1.0
Kies	0	2.82	3.87	4.09	4.12	4.12
Brown	0	3.00	3.83	3.98		
the above	0	1.74	3.14	3.88	4.14	4.0
($\times 1.12$)		(1.95)	(3.41)			



The differences between the present work and Kies' result is quite small as $\frac{a}{w} \geq 0.5$.

Chebyshev polynomials: (for complicated applied stress)

$$T_0(\omega) = 1, \quad T_1(\omega) = \omega, \quad T_2(\omega) = 2\omega^2 - 1, \quad T_3(\omega) = 4\omega^3 - 3\omega \dots$$

$$U_0(\omega) = 0, \quad U_1(\omega) = 1, \quad U_2(\omega) = 2\omega, \quad U_3(\omega) = 4\omega^2 - 1 \dots$$

$$\frac{1}{\pi} \int_{-1}^1 \frac{1}{y-\omega} \left[\frac{T_n(y)}{(1-y^2)^{\frac{n}{2}}} \right] dy = U_n(\omega)$$

\downarrow

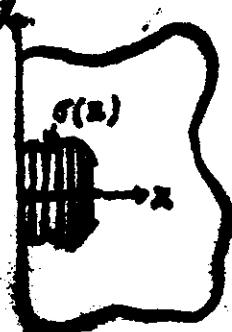
$D(\frac{y}{a}) \qquad \sigma_{yy}^n(a)$

$\xrightarrow{K^2}$

$$A \int_{-1}^1 \frac{D(\frac{y}{a}) \sigma(\frac{y}{a})}{(\frac{y}{a}) - \frac{z}{a}} dy = \sigma_{yy}^n(\frac{z}{a})$$

$$\sigma_{yy}^n(\omega) \rightarrow \sum C_n(\omega) U_n(\frac{z}{a})$$

$$D(\frac{z}{a}) = \frac{1}{4\pi} \sum_n \frac{C_n(\omega) T_n(\frac{z}{a})}{(1 - (\frac{z}{a})^2)^{\frac{n}{2}}}$$

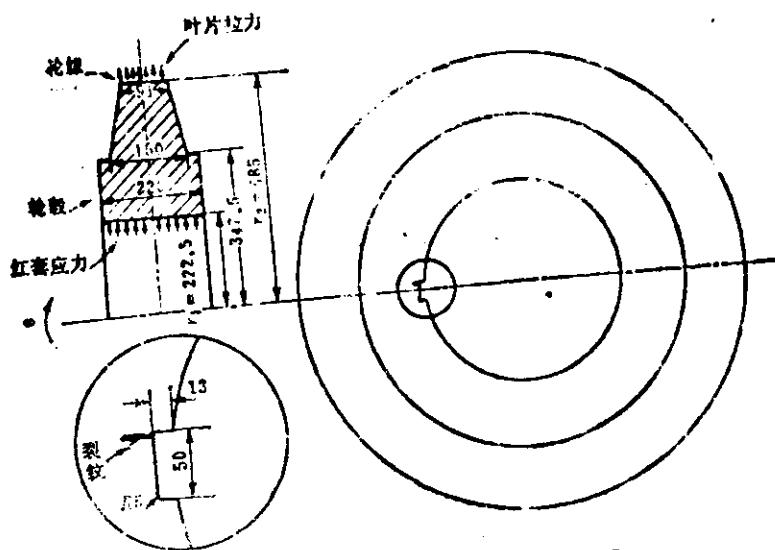


for edge crack in an infinitely large sheet.

$$y, \quad \sigma(z) = \sigma \sum_{\text{ordinary solutions}} C_n(\omega), \quad K^2 = \sigma C_0 F(\omega) \sqrt{\omega},$$

$$F(\omega) = 1.12 + 0.56 \left(\frac{\omega}{\omega_0} \right) + 0.56 \left(\frac{\omega}{\omega_0} \right)^2 + 0.42 \left(\frac{\omega}{\omega_0} \right)^3$$

$$F(\omega) = 1.12 + 0.677 \left(\frac{\omega}{\omega_0} \right) + 0.52 \left(\frac{\omega}{\omega_0} \right)^2 + 0.438 \left(\frac{\omega}{\omega_0} \right)^3 \quad \begin{matrix} \text{Handbook} \\ \text{SMAC, (1973)} \end{matrix}$$

图6-1 A π -25-2型第14级叶轮示意图

2.5~3. 1°机组自1952年投入运行至1975年发现裂纹，累积运行15.4万小时，工作情况正常。该叶轮材料金相组织为回火索氏体。化学成分和常规力学性能见表1和表2。

表1 材料的化学成分 Wt%

C	Si	Mn	Cr	Ni	Mo	P	S
0.33	0.30	0.47	0.83	2.02	0.31	0.014	0.018

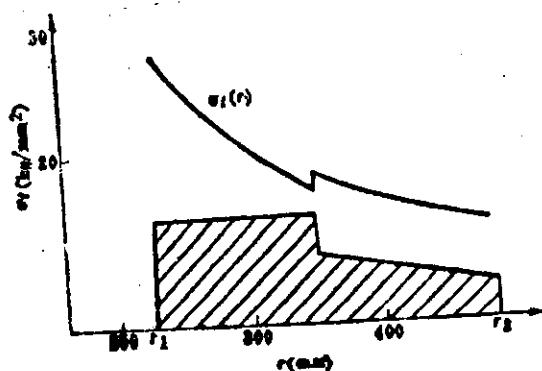
图6-2 A π -25-2型第1级叶轮正常运转时切应力沿径向分布图

表2 材料的常规力学性能

指标	σ_{s1} (kg/mm²)	σ_s (kg/mm²)	δ_5 (%)	ψ (%)	a (Kg·m/cm²)	FATT (°C)
抗拉强度	72.8	90.2	18.0	54.8	7.0	15
抗弯强度	66.8	87.0	20.3	59.1	8.8	35

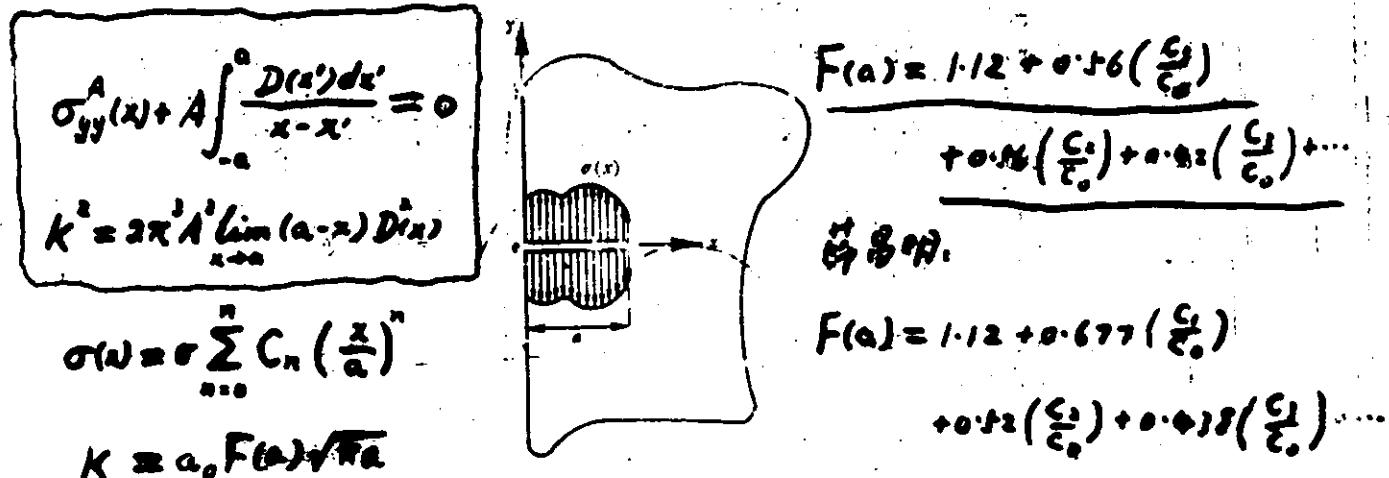


Fig. 2. A single edge crack under arbitrary distribution of applied stresses in a semi-ininitely large plate.

percentage are about 2—15%. The calculation based on the dislocation theory is simpler (Fig. 3).

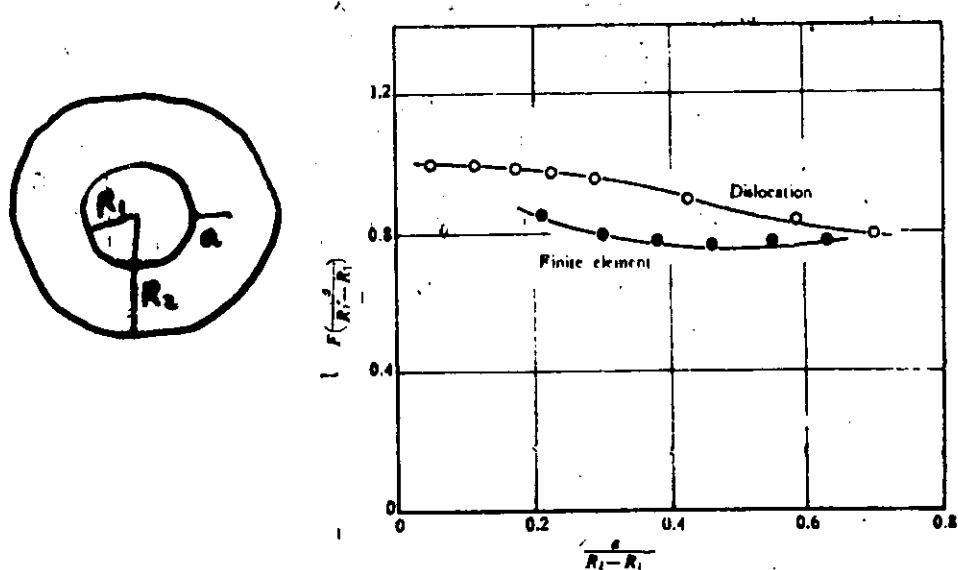


Fig. 3. The stress intensity factors of a rotary plate with an edge crack at the central hole.

The Chebynev polynomials may be applied to more general cases. Any function $\sigma(x)$ may be considered as a "vector" in Hilbert space. The orthogonal fundamental function sets (such as $U_n(x)$ or $T_n(x)$) are the base vectors in Hilbert space. If the orthogonality and weight functions are known, the components of every base vector (the coefficient of certain $U_n(x)$ terms in the expansion of the function) can be calculated by the expansion of $\sigma(x)$. So any stress distribution function $\sigma(x)$ can be expressed in terms of $U_n(x)$ in principle.

该材料的: 抗拉强度 3000 kg/mm^2 , $K_{Ic} = 300 \text{ kg/mm}^{3/2}$

$a_c = 60 \sim 70 \text{ mm}$

For a rotary plate with an edge crack at the central hole.

$$\frac{\Delta K}{K} \approx 2 - 1.5 \% \quad (\text{in comparison with finite element method})$$

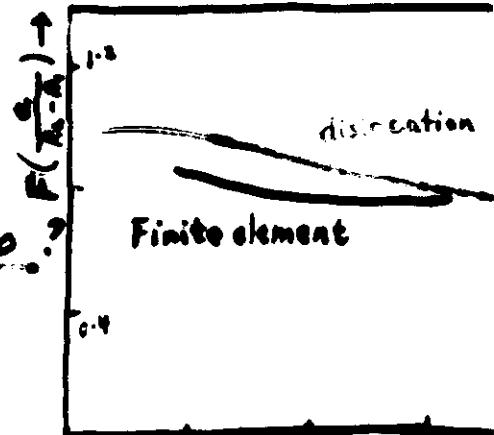
For ^{an} actual steam turbine wheel. ($f = 3000 \text{ rev/min.}$)

$$K_{ic} = 380 \text{ kN/mm}^{0.5} \quad (\text{experimental results})$$

$a(\text{mm})$	1	2	3	4	5	6	7
$K(\text{kN/mm})$	160	230	277	314	344	371	394

$$a_c \approx 6.0 \approx 70 \text{ mm}$$

Is there any advantage of doing so?



1. Wide application possible

$\sigma_y(a)$ (crack-free material) \rightarrow data of design $\frac{a}{b-a} \rightarrow 0.6$
 \rightarrow experimental measurements (Kinsler
Act. Mech. Stress, 2(1978)192.)

2. It would save us much time and labour of calculation.

Many engineering problems require prompt solution rather than accuracy.

