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INTERNATIONAL CENTRE FOR SCIENCE AND HIGH TECHNOLOGY

c/o INTERNATIONAL CENTRE FOR THEORETICAL PHYSICS 34100 TRIESTE (ITALY) VIA GRIGNANO 9 (ADRIATICO PALACE) P.O. BOX 586 TELEPHONE 0432/201471 TELEFAX 0432/201475 TELEX 40049 APH I

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**"College on Atmospheric Boundary Layer
and Air Pollution Modelling"**
16 May - 3 June 1994

"A 4-layer Model of Turbulent Air Flow over Low Hills"
(The Jackson Hunt Approach)

S.E. BELCHER
Department of Meteorology
University of Reading
Reading, United Kingdom

Please note: These notes are intended for internal distribution only.

A 4 Layer Model of Turbulent Air Flow over

Low Hills

(The Jackson and Hunt Approach)

S.E. Belcher

Department of Meteorology
University of Reading

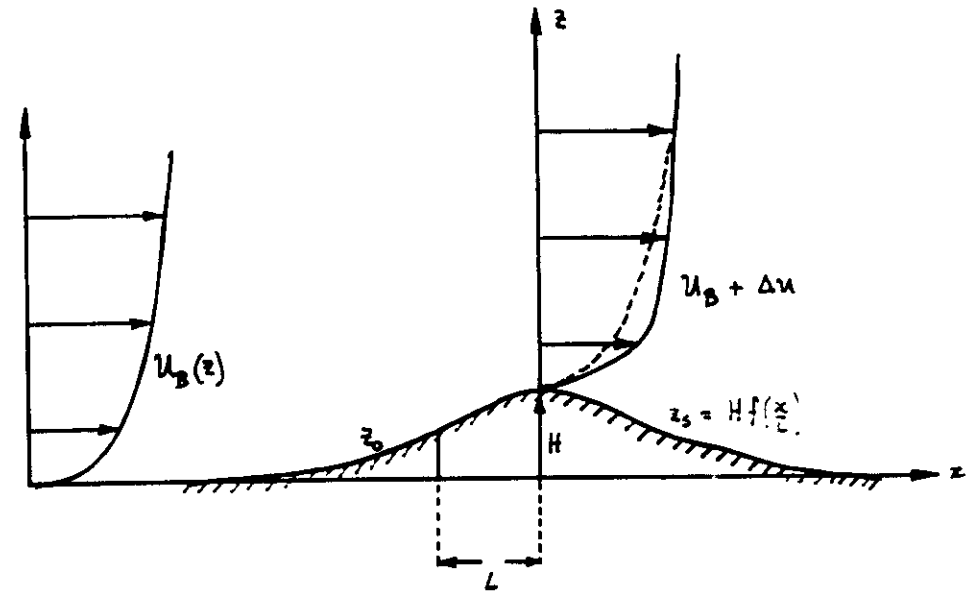
We shall develop a method for analysing how a turbulent boundary layer responds to changes in surface conditions

eg. a hill; a change in surface roughness; a water wave.

Qualitative Questions:

1. How can these flows be modelled?
2. How much is the mean flow changed; where is it changed most?
3. How does the turbulence change?

Flow Geometry



Approach Flow: $U_B(z) = \frac{u_*}{\kappa} \ln z/z_0$

$$\tau_B(z) = \rho u_*^2$$

⇒ 2 key parameters: u_* and z_0

Also have boundary layer depth δ

($\delta \sim 1\text{km}$, $z_0 \sim 0.1\text{m}$, $u_{*0} \sim 10\text{--}20\text{ms}^{-1}$, $u_* \sim 0.3\text{--}0.8\text{ms}^{-1}$)

Hill:

Two dimensional $z_s = H f(x/L)$

Low slope, ie. $H \ll L$

⇒ $|\Delta u| \ll U_B$

Then can analyse linear changes to U_B

Mathematical Model

Write flow variables on hill as approach flow plus perturbation

$$u = u_0 + \Delta u, \quad w = \Delta w,$$

$$p = p_0 + \Delta p, \quad T = T_0 + \Delta T$$

Substitute into Reynolds-averaged Navier-Stokes equations and linearise (i.e. neglect products of perturbations)

$$\rho \left\{ u_0 \frac{\partial \Delta u}{\partial x} + \Delta w \frac{d u_0}{dz} \right\} = - \frac{\partial \Delta p}{\partial x} + \frac{\partial \Delta \tau_{xz}}{\partial z} + \cancel{\frac{\partial \Delta \tau_{xz}}{\partial x}}$$

$$\rho u_0 \frac{\partial \Delta w}{\partial x} = - \frac{\partial \Delta p}{\partial z} + \frac{\partial \Delta \tau_{xz}}{\partial x} + \cancel{\frac{\partial \Delta \tau_{xz}}{\partial z}}$$

$$\frac{\partial \Delta u}{\partial x} + \frac{\partial \Delta w}{\partial z} = 0$$

These are p.d.e.s. To solve them:-

1. Take Fourier Transform in x : $\Delta \tilde{u} = \int_{-\infty}^{\infty} \Delta u e^{-ikx} dx$

Then p.d.e.s $\mapsto$ o.d.e.s because $\partial/\partial x \mapsto ik$

$\Rightarrow$ Fundamental solutions are for cosinusoidal hill.

2. Need a model for Reynolds stress tensor

$$\Delta \tau_{ij} = - \rho \Delta \overline{u_i' u_j'}$$

In fact only $\Delta \tau_{xz} \equiv \Delta \tau$ is dynamically significant.

Preliminary Scaling: Try to simplify equations

$$T = T_0 + \Delta T = \rho u_*^2 + \Delta T = \rho u_*^2 \left(1 + \frac{\Delta T}{\rho u_*^2} \right)$$

$$\text{We expect } \Delta T \sim \frac{H}{L} \rho u_*^2$$

$$\text{In } x\text{-momentum eqn. we have: } \frac{\partial \Delta T}{\partial z} \sim \frac{H/L \rho u_*^2}{L}$$

$$\text{also have inertial terms: } \rho u_0 \frac{\partial \Delta u}{\partial x} \sim \rho u_0 \frac{H/L u_0}{L}$$

$\therefore$ compare these terms:

$$\frac{\partial \Delta T / \partial z}{\rho u_0 \partial \Delta u / \partial x} \sim \frac{u_*^2 / L}{u_0^2 / L} = \frac{u_*^2}{u_0^2} \ll 1$$

$$\text{Typically, in atmosphere } \frac{u_*}{u_0} \sim 0.03 - 0.07$$

So we can ignore the Reynolds stress gradients?

$$x\text{-momentum: } \rho \left\{ u_0 \frac{\partial \Delta u}{\partial x} + \Delta w \frac{d u_0}{dz} \right\} = - \frac{\partial \Delta p}{\partial x} + \cancel{\frac{\partial \Delta \tau}{\partial x}}$$

Inviscid irrotational flow

We have ignored Reynolds stress gradients, now also ignore shear in approach flow, $du/dz = u_x/kz$

ie. $\frac{du}{dz} \approx 0$, $u \approx u_0$ constant. When z is big

$$\Rightarrow \rho u_0 \frac{\partial \Delta u}{\partial x} \approx - \frac{\partial \Delta p}{\partial x}$$

$$\rho u_0 \frac{\partial \Delta w}{\partial z} \approx - \frac{\partial \Delta p}{\partial z}$$

$$\frac{\partial \Delta u}{\partial x} + \frac{\partial \Delta w}{\partial z} = 0$$

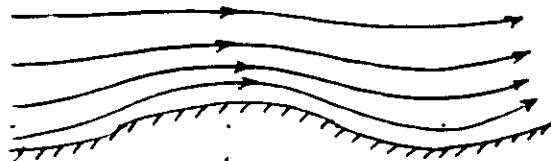
Boundary conditions:

(i) at surface, $z = H f(\frac{x}{L})$: $\Delta w(0) = u_0 \frac{\partial z}{\partial x}$
ie streamline follows surface $= u_0 \frac{H}{L} f'$

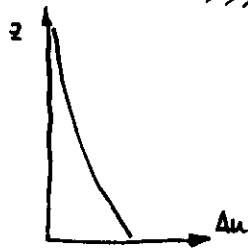
(ii) High above hill, $z \rightarrow \infty$: $\Delta w \rightarrow 0$

For cosine hill $f = \cos 2\pi x/L$ solution is

$$\left. \begin{aligned} \Delta u &= u_0 e^{-2\pi z/L} \cos 2\pi x/L \\ \Delta w &= -u_0 e^{-2\pi z/L} \sin 2\pi x/L \end{aligned} \right\}$$

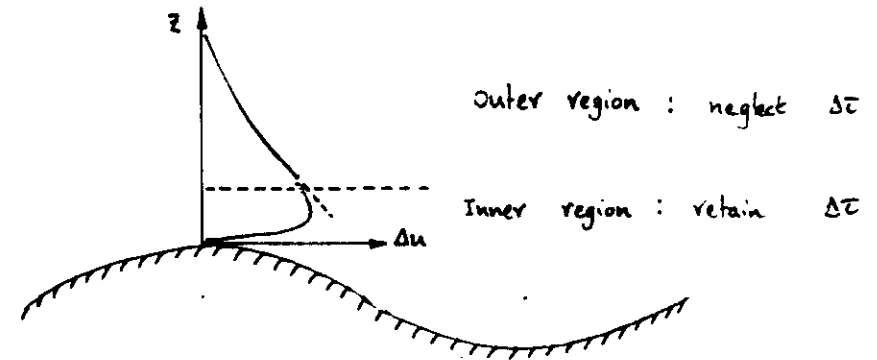


streamlines



But Δu does not satisfy the b.c. $\Delta u = 0$ at surface!
 $\Rightarrow$ Need to bring back $\partial \Delta u / \partial z$ near surface.

The need for inner and outer regions:-



1. Outer region

The Reynolds stress gradient can be neglected over most of the flow. The potential flow solⁿ is valid through most of this region.

2. Inner Region

Within a narrow region close to the surface, the velocity perturbation is reduced to zero at surface.

This is mediated by Reynolds stress gradients in the inner region.

The inner region is like a boundary layer that lies within the upstream boundary layer, and so the inner region is sometimes called an internal boundary layer.

$\Rightarrow$ We need a model for $\Delta \epsilon$.

Effects of hill on turbulence

1. Consider approach flow, ie no hill.

In surface layer: $\tau_B = K^2 z^2 \left| \frac{\partial u_B}{\partial z} \right| \frac{\partial u_B}{\partial z}$

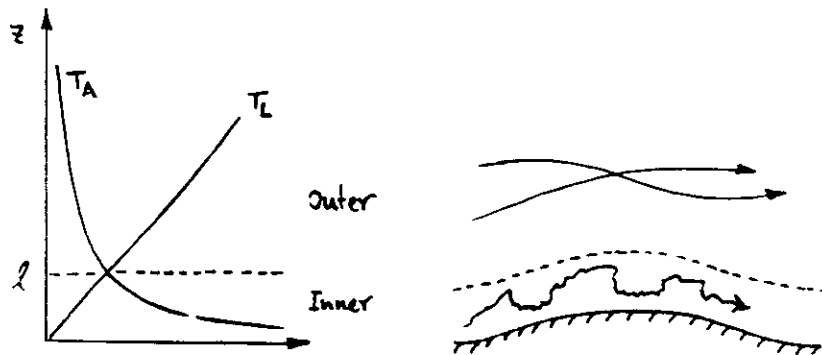
mixing-length model is a good model.

but care needed for $\Delta\tau$

2. Stress perturbation $\Delta\tau$

2 timescales: $T_A = L/u_B(z)$ = advection timescale

$T_L = z/u_*$ = eddy turnover timescale.



$T_A \sim T_L$ when $z \sim l$ where $l \ln l/z_0 \sim L$

Then $\frac{l}{L} \sim \frac{u_*}{u_0} \ll 1$

Turbulence Models

Inner Region: when $z \ll l \Rightarrow T_L \ll T_A$

$\Rightarrow$ Turbulence in local equilibrium

$\Rightarrow$ Can use mixing length model:

$$\tau + \Delta\tau = K^2 z^2 \left(\frac{du_B}{dz} + \frac{\partial \Delta u}{\partial z} \right)^2$$

$$\Rightarrow \Delta\tau = 2Kz u_* \frac{\partial \Delta u}{\partial z}$$

Outer Region: when $z \gg l \Rightarrow T_A \gg T_L$

$\Rightarrow$ changes to turbulence are rapid

$\Rightarrow$ can compute changes using rapid distortion theory

RDT shows that $\Delta\tau \sim O(u_*^2)$ as used above.

Hence stress perturbations are indeed negligible in outer.

N.B. If mixing length is used in outer region then

$$\Delta\tau = O(u_* u_0) = O\left(u_*^2 \frac{u_0}{u_*}\right)$$

ie too big by factor of $u_0/u_* \gg 1$.

This does not affect the mean flow, but is important for $\Delta\tau$ calculations and also affects force on hill.

Overall structure of the perturbed boundary layer

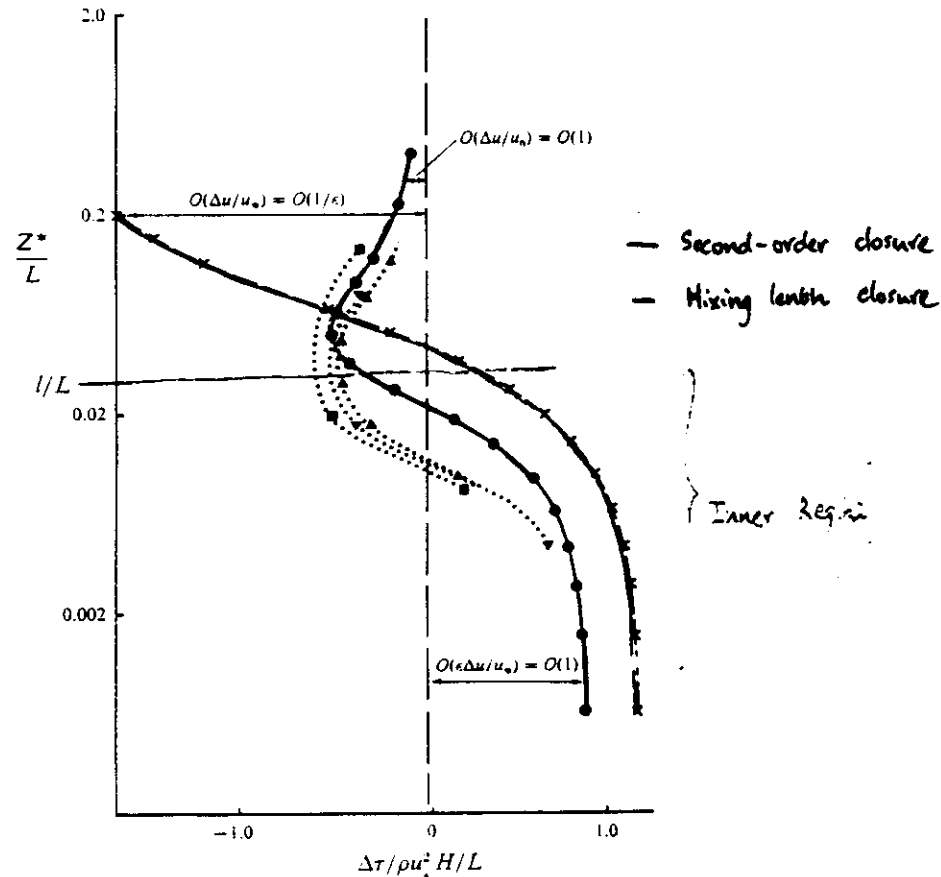
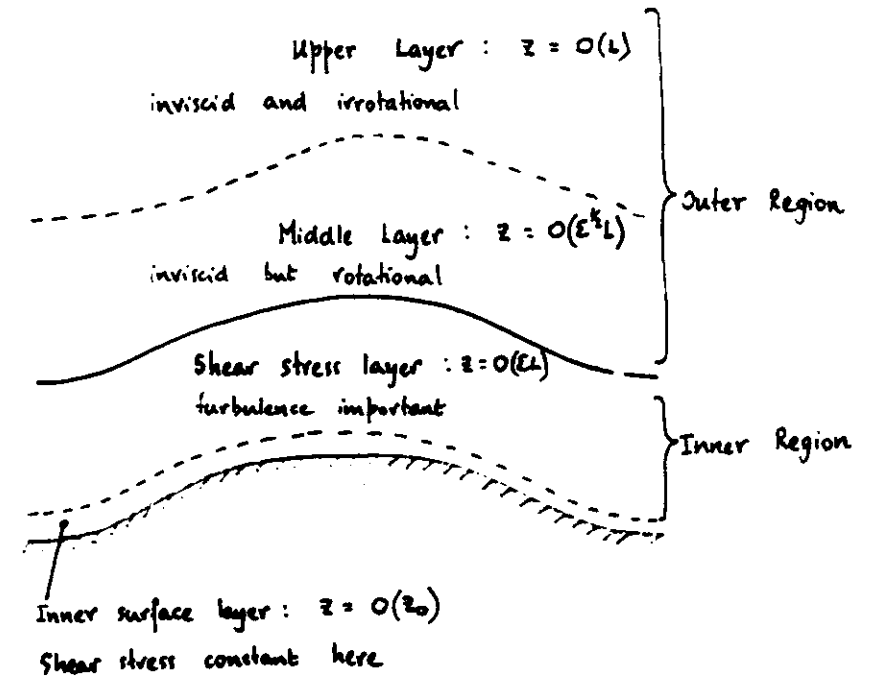


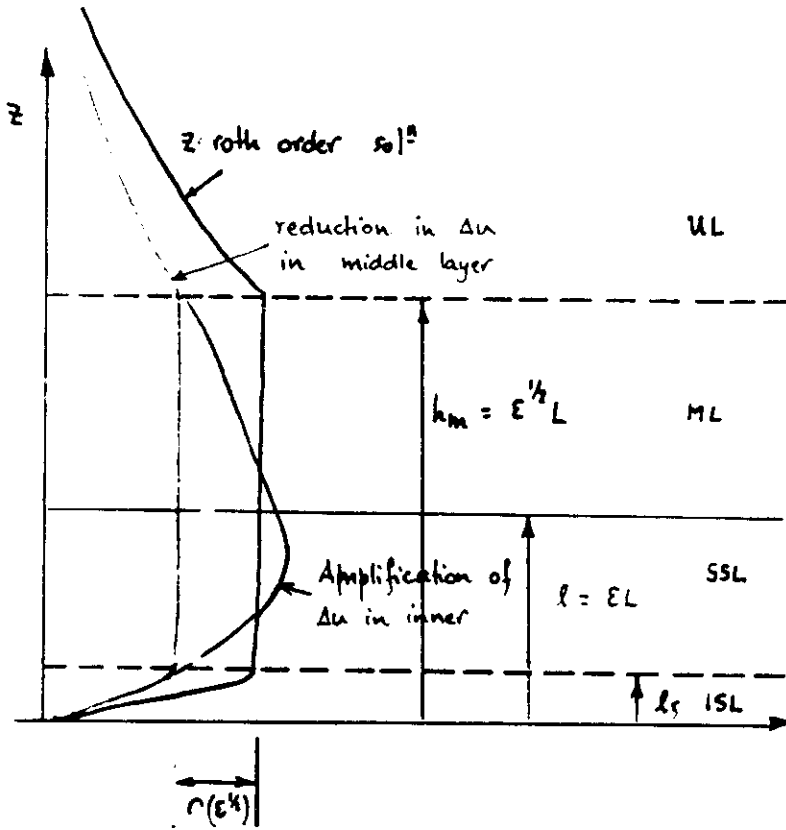
FIGURE 3. Values of the shear-stress perturbations at the crest of an isolated hill computed using the second-order-closure and mixing-length models (with the same hill shape and parameters as the numerical results in figure 4c), together with comparisons with field data. Note how the second-order closure model, which is in good agreement with the field data, shows a small perturbation in the outer region, whereas the mixing-length model erroneously predicts a large negative perturbation: —, upwind profile; —○—, numerical results using the second-order-closure model; —x—, numerical results using the mixing-length model; ▲, field data for Askervein (reported in Zeman & Jensen 1987); ▼, field data for Nyland hill (Mason 1986b); ■, field data for Blashaval (Mason & King 1985).



ie 4 layers!

Solutions have been coded up into FLOWSTAR
which calculates air flow over arbitrarily shaped hills.

schematic of solution at crest of hill



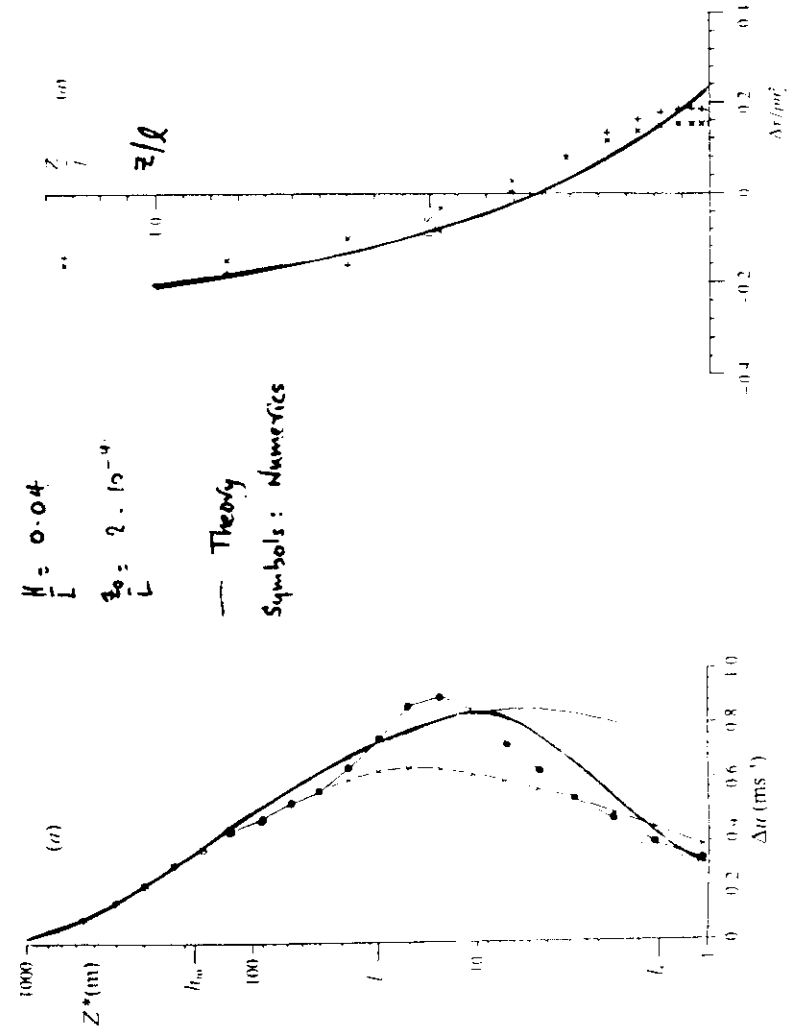
$O(\epsilon)$

If $L = 250m$ and $z_0 = 0.1m$
then

$$h_m = 95m$$

$$l = 16m$$

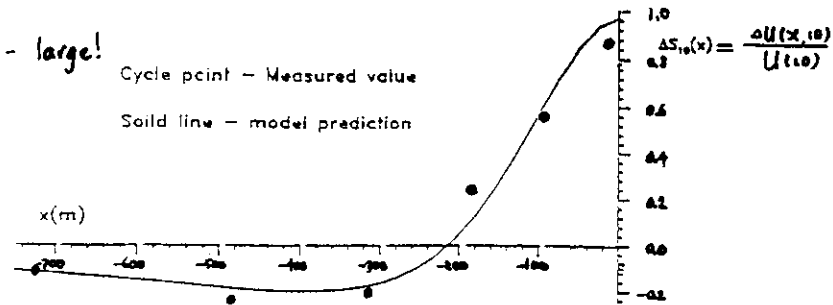
$$L_s = 13m$$



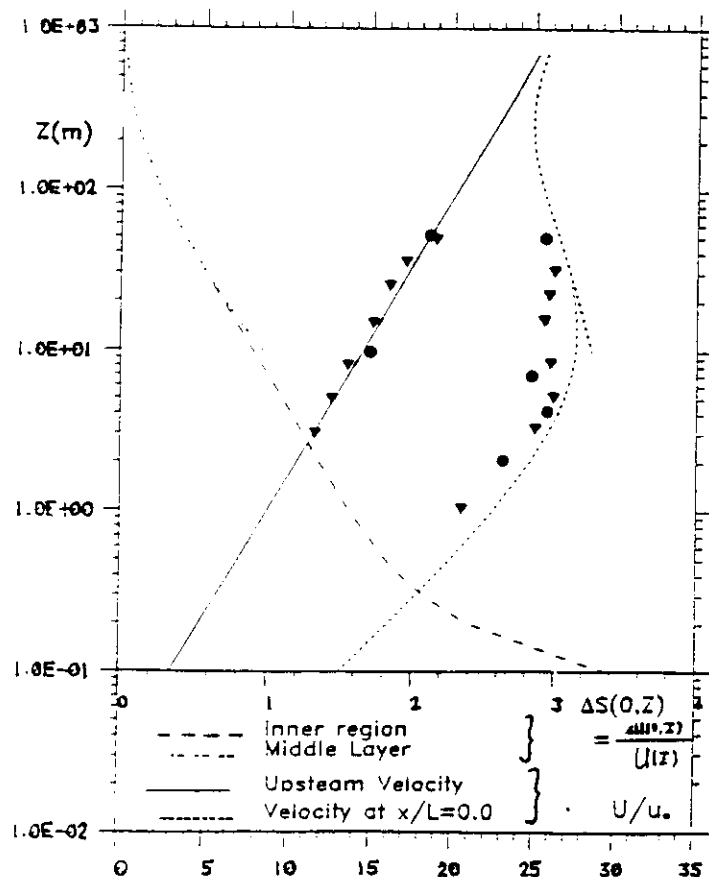
Comparison of the theory with a near-linear model with second-order closure (Newley 1986). Crest of sinusoidal terrain

Comparison with field data from Arkenvein experiment

$$\frac{H}{L} = 0.3 - \text{large!}$$



Speed-up at $Z=10m$ along the x direction



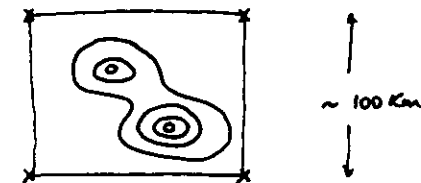
Force on the hill

Turbulent flow over a flat surface

⇒ frictional force on the surface.

When surface has undulations = hills force increased

Motivation: Large scale weather forecast models

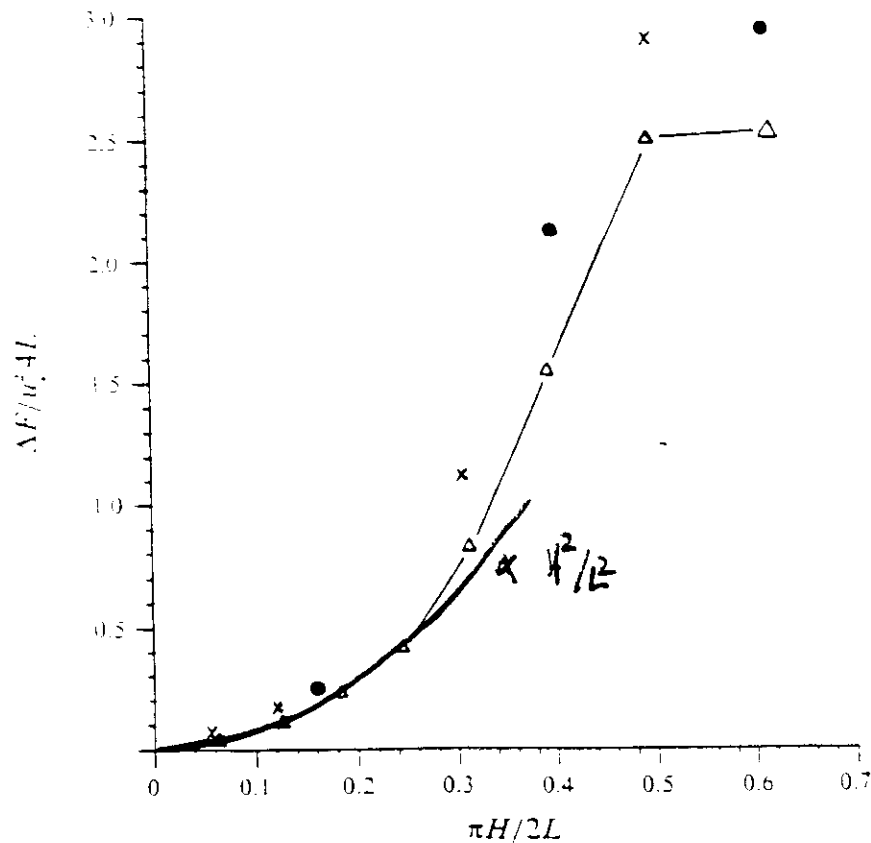


Effect of hills on momentum exchange must be parameterised

$$\Delta F = \frac{1}{L} \int_{-\infty}^{\infty} \Delta \tau f' dx + \frac{1}{L} \int_{-\infty}^{\infty} \Delta \tau dx \quad \text{smaller}$$

Theory gives
$$\Delta F = U_*^2 \beta \frac{1}{L} \int_{-\infty}^{\infty} \{f'\}^2 dx$$

$$\beta = \frac{U_B(\varepsilon^k L)}{U_B(\varepsilon L)}$$



Variation of drag with slope

Review of requirements for theory to be valid

1. Low slope, $H/L \ll 1$

Then the governing equations can be linearised

Typically okay up to crest of hill if $H/L \lesssim 1/3$

2. $\varepsilon = u_* / u_0 \ll 1$

Then the Reynolds stress gradients can be neglected except in the inner region

In practice this can be determined by the ratio z_0/L

$$\frac{u_*}{u_0} \sim \frac{1}{1/2 \ln 4/z_0}$$

3. Logarithmic velocity profile describes u_y in inner region

This is used explicitly in the solution procedure.

In the outer region u_y can have a more general form.

$u_y(z)$ logarithmic if $\varepsilon \lesssim 3/5$

$$\therefore \text{Require } \varepsilon \lesssim 3/5 \Rightarrow L \lesssim \frac{1}{5} \delta \frac{u_0}{u_*} \sim 20 \text{ km}$$

Brief Bibliography of papers using the 4 layer method.

1. Flow over hills

Jackson & Hunt (1975) Quart. J. Roy. Met. Soc. 101 929 - 955

Sykes (1980) J. Fluid Mech. 101 647 - 670

Hunt, Leiberman & Richards (1988) Q. J. R. Met. Soc. 114 1435 - 1471

Carruthers & Hunt (1990) In: Atmospheric processes over complex terrain

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Belcher, Newley & Hunt (1993) J. Fluid Mech. 249 557 - 596

2. Flow over roughness changes

Belcher, Xu & Hunt (1990) Q. J. R. Met. Soc. 116 611 - 635.

3. Flow over moving ocean-waves

Belcher & Hunt (1993) J. Fluid Mech. 251 109 - 148

Belcher, Harris & Street (1994) J. Fluid Mech. to appear vol. 271